

UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION

## OPERATION LIMITATIONS

ENGINE	MAKE <b>P &amp; W</b>	AIRCRAFT	MAKE <b>GRUMMAN</b>	NATIONALITY AND REGISTRATION MARKS <b>U.S.A. NC-1048V</b>	
	MODEL <b>R 985-AN3</b>		MODEL <b>G 21A</b>	SERIAL NO. <b>37793</b>	TYPE CERTIF. NO. <b>654</b>

## ENGINE AND AIR SPEED LIMITS NOT TO BE EXCEEDED

(All Values Are Maximum and Are NOT RECOMMENDED OPERATING LIMITS)

ENGINE LIMITS							TRUE INDICATED AIR SPEED		
	MINUTES	ALTITUDE	IN. HG.	R. P. M.	HP.	MIN. OCT.		M. P. H. LAND	M. P. H. SEA
TAKE-OFF	1	ANY	36.5	2300	450	91	CLIMB OR LEVEL FLIGHT	194	194
SEA LEVEL		TO	34.5	2200	400	91	GLIDE OR DRIVE (Smooth Air Only)	225	225
ALTITUDE	FROM	5000	33.5	2200	400	91	FLAPS EXTENDED	110	110

\*METO—MAXIMUM EXCEPT TAKE-OFF

TAKE-OFF WEIGHT			LANDING WEIGHT		
LAND	8000	SEA	8000	LAND	8000

## C. G. LIMITS

LAND	+20.5 to +33	SEA	+20.5 to +33
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## WEIGHT

SEE WEIGHT AND BALANCE DATA FOR LOADING INFORMATION

## DATUM

Wing leading edge at fuselage

## LIMITATIONS

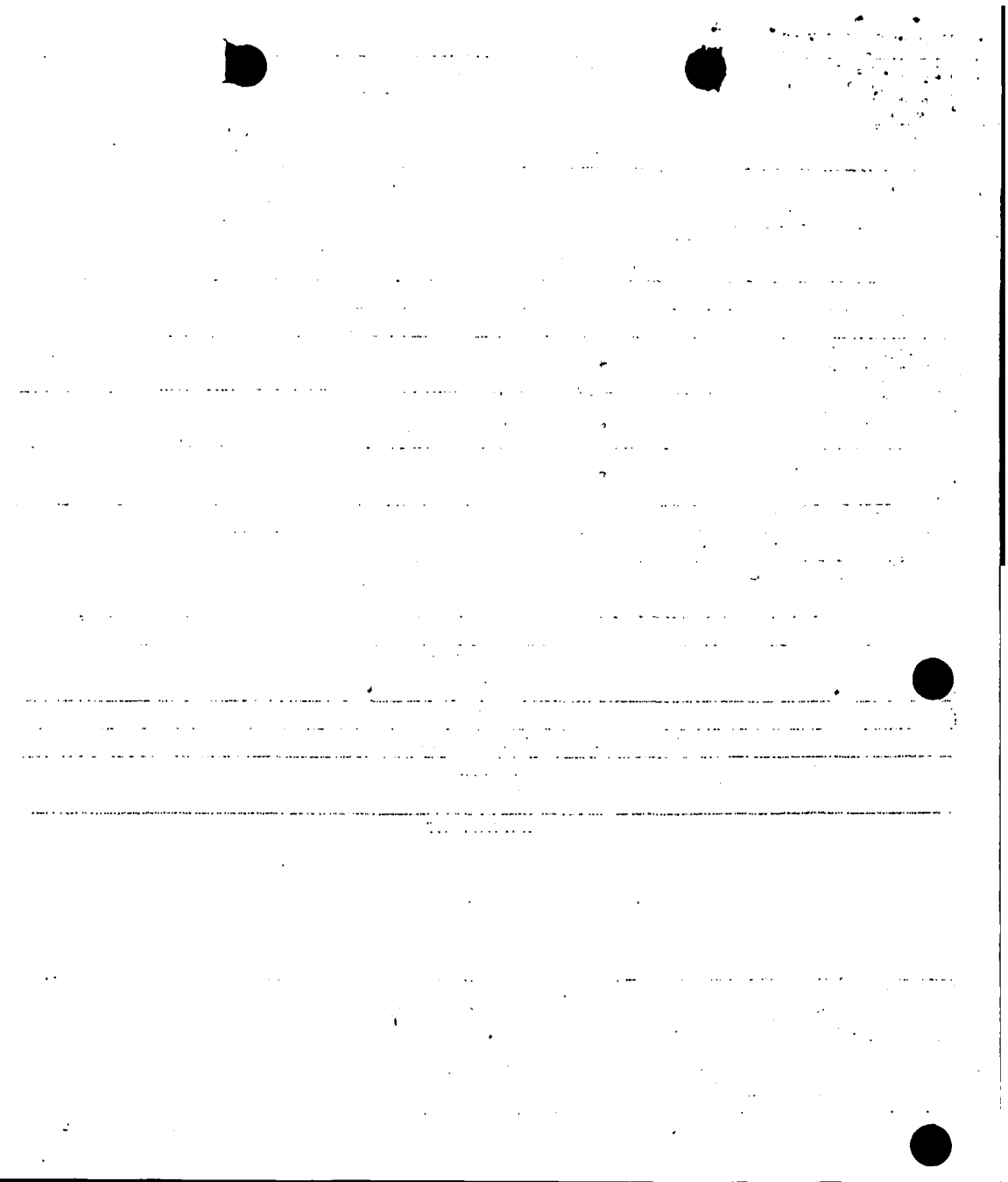
INSPECTOR'S OR DESIGNEE'S SIGNATURE <b>H.E. Waddt</b>	DESIG. NO. <b>9741</b>	DATE <b>July 16, 1949</b>
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ADDITIONAL LIMITATIONS

YES ☐NO ☒

(IF YES—SEE OVER)

THIS FORM MUST BE AVAILABLE IN THE AIRCRAFT WHEN OPERATED



APPLICATION FOR AIRWORTHINESS CERTIFICATE AND/OR  
ANNUAL INSPECTION OF AN AIRCRAFT

INSTRUCTIONS  
Please submit this form to the Civil Aeronautics Administration Field Representative.

APPLICATION (Check whether) <input type="checkbox"/> ORIGINAL AIRWORTHINESS CERTIF. <input checked="" type="checkbox"/> ANNUAL INSPECTION	CAA IDENTIFICATION <input checked="" type="checkbox"/> NC <input type="checkbox"/> NR <input type="checkbox"/> OTHER (Specify) _____ <input type="checkbox"/> NX <input type="checkbox"/> NL
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AIRCRAFT

MAKE <b>GRUMMAN</b>	MODEL <b>G 21A</b>
REGISTRATION NO. <b>NC1048V</b>	MANUFACTURER'S SERIAL NO. <b>37793</b>
DATE MANUFACTURED	TYPE CERTIFICATE NO. <b>654</b>

ENGINE

MAKE <b>Pratt &amp; Whitney</b>	MODEL <b>R 985 AN3</b>
OWNER'S NAME <b>Arthur James Williams</b>	PERMANENT ADDRESS (Street and number, City, Zone and State) <b>32 Main &amp; Hope Streets, Georgetown, British Guiana, S. A.</b>

ATTACHMENTS (Check which) <input checked="" type="checkbox"/> ACA-319 <input type="checkbox"/> WEIGHT AND BALANCE REPORT <input type="checkbox"/> ACA-337 <input type="checkbox"/> SPECIAL APPROVALS <input type="checkbox"/> ACA-805 <input type="checkbox"/> DATA, DRAWINGS, ETC. <input type="checkbox"/> ACA-317 <input type="checkbox"/> UNAPPROVED DEVIATION DATA	I CERTIFY THAT THE ABOVE STATEMENTS ARE TRUE.  <b>James Bow</b> MAINTENANCE AUTHORIZED AGENT <b>July 29, 1950</b> <b>Maintenance Superintendent</b> DATE TITLE
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FORM ACA-305a (FORMERLY ACA-307)

AIRCRAFT INSPECTION REPORT

(To be completed by a CAA inspector or a designated inspector or representative)

IT HAS BEEN DETERMINED THAT THE AIRCRAFT DESCRIBED IN 305 ABOVE IS IN CONFORMITY WITH THE FOLLOWING

ALL APPLICABLE MANDATORY NOTES, INSTRUMENT MARKINGS AND PLACARDING REQUIREMENTS HAVE BEEN COMPLIED WITH ☒ YES ☐ NO

(AIRCRAFT SPECIFICATION - AIRWORTHINESS DIRECTIVE, NO(S). A- **654-3 through 48-18-1**  
(SPECIFY)

FORM ACA-1362, CERTIFICATE OF AIRWORTHINESS, ISSUED ☐ ORIGINAL ☒ ANNUAL INSPECTION

(Check whether)

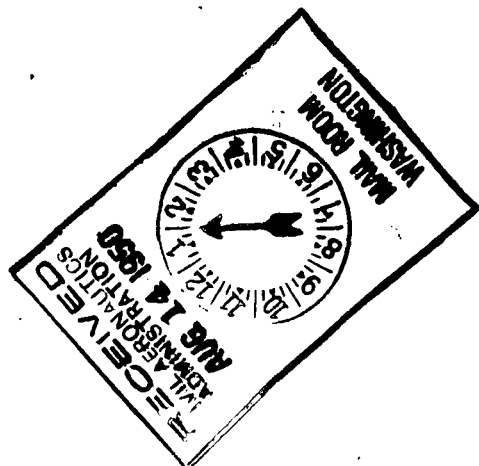
☐ OPERATION LIMITATIONS FORM ACA-309 WAS ISSUED, OR  
☒ APPROVED AIRPLANE FLIGHT MANUAL IS IN THE AIRCRAFT **is available in aircraft**

FINDINGS

<input checked="" type="checkbox"/> AIRWORTHY <input type="checkbox"/> UNAIRWORTHY	DESIGNEE'S SIGNATURE AND NO. <b>H. E. Wendt 9741</b>	DATE <b>July 29/50</b>
	CAA INSPECTOR'S SIGNATURE <b>James E. Curran</b>	<input type="checkbox"/> ACCEPTED <input type="checkbox"/> REINSPECTED DATE <b>8-7-50</b>

REASON FOR DISAPPROVAL, OR REMARKS (Indicate if you have used the reverse to continue this or other item ☐ Yes ☒ No)

AUG 22 9 40 AM '50  
RECEIVED  
CERTIFICATE SECTION



P. O. Box 52  
Miami Springs, Fla

August 9, 1950

International Region

1048 V

TO : Administrator, International Region TR-500  
Washington, D. C. Attn: CESA Specialist

FROM : Chief Advisor, IBO - Miami, Fla

SUBJECT: Form ACA 305 - Application for Airworthiness Certificate  
and/or Inspection of an Aircraft

Enclosed is an executed Form ACA 305 covering annual inspection  
of a Grumman 321A aircraft, NC1046V, owned by Arthur James  
Williams and inspected by Mr. H. E. Wendt, DAI #9741, July 29,  
1950.

The above is forwarded for your further handling.

Original signed by  
Clyde S. Angel for

Frank G. Stone

3  
Encl.

NOTICE OF RECEIPT  
RECEIVED  
APR 22 9 43 AM '77  
U.S. AIR FORCE

Post Office Box 102  
Georgetown  
British Guiana, S.A.  
August 3rd, 1950

Chief Advisor  
International District Office  
Post Office Box 52  
Miami Springs, Florida, U.S.A.

Dear Sir,

Enclosed please find Forms 305 and 305A covering  
Grumman G-21A, NC-1048V.

Your letter of July 25th received and I note that  
Airworthiness Directive 48-17-1 is considered not mandatory,  
but recommended. Delivery on the parts for compliance is  
fairly certain now during the current month and the installation  
will be done as soon as the parts arrive.

Sincerely yours,

Original signed by

H. E. Wendt. DAMI 9741

HEW/rp  
enc:

RECEIVED  
GENERAL INVESTIGATIVE SECTION  
FBI  
JUN 22 9 40 AM '68



Administrator, International Region IR-500

October 21, 1949

Attention: USIA Specialist

DE-575

Chief Advisor

IFO - Miami, Fla

Application for Airworthiness Certificate for Grumman C21A, Georgetown,  
British Guiana

As per request contained in your letter of October 12, 1949, the  
attached forms, covering certification of subject aircraft have  
been processed and are being returned for filing with A-300, Air-  
craft Record Branch.

It is requested that we be advised if there are any additional  
Maintenance Designees in our assigned area and, if there are, it  
is believed that the files on these personnel, along with Mr.  
Wendt's, should be forwarded this office in order that proper  
supervision can be maintained.

We will withhold notifying Mr. Wendt to communicate with this of-  
fice until a reply is received from you.

Original signed by

F. C. Stone

CCGibson/es

Encl.

RECEIVED  
JAN 11 1969

APPLICATION FOR AIRWORTHINESS CERTIFICATE AND/OR  
ANNUAL INSPECTION OF AN AIRCRAFTINSTRUCTIONS  
Please submit this form to the Civil Aero-  
nautics Administration Field Representative.

## APPLICATION (Check whether)

☐ ORIGINAL AIRWORTHINESS CERTIF.☒ ANNUAL INSPECTION

## CAA IDENTIFICATION

☒ NC☐ NR☐ OTHER (Specify) \_\_\_\_\_☐ NX☐ NL

## AIRCRAFT

MAKE

GRUMMAN

MODEL

G21A

REGISTRATION NO.

NC1048V

MANUFACTURER'S SERIAL NO.

37793

DATE MANUFACTURED

TYPE CERTIFICATE NO.

654

## ENGINE

MAKE

PRATT &amp; WHITNEY

MODEL

R 985-AN3

OWNER'S NAME

Arthur James Williams

PERMANENT ADDRESS (Street and number, City, Zone and State)

32 Main & Hope Streets, Georgetown  
British Guiana

## ATTACHMENTS (Check which)

☐ ACA-319☒ WEIGHT AND BALANCE REPORT☒ ACA-337☐ SPECIAL APPROVALS☐ ACA-805☐ DATA, DRAWINGS, ETC.☐ ACA-317☐ UNAPPROVED DEVIATION DATA

I CERTIFY THAT THE ABOVE STATEMENTS ARE TRUE.

James Bow

OWNER OR AUTHORIZED AGENT

July 16, 1949

DATE

Maintenance Supt.

TITLE

FORM ACA-305a (FORMERLY ACA-307)

## AIRCRAFT INSPECTION REPORT

(To be completed by a CAA inspector or a designated inspector or representative)

IT HAS BEEN DETERMINED THAT THE AIRCRAFT DESCRIBED IN 305 ABOVE IS IN CONFORMITY WITH THE FOLLOWING

ALL APPLICABLE MANDATORY NOTES, INSTRUMENT MARKINGS AND PLACARDING REQUIREMENTS HAVE BEEN COMPLIED  
WITH ☒ YES ☐ NO(AIRCRAFT SPECIFICATION - AIRWORTHINESS DIRECTIVE, NO(S). A-654-3 A.D. complied with  
(SPECIFY)

FORM ACA-1362, CERTIFICATE OF AIRWORTHINESS, ISSUED

☐ ORIGINAL☒ ANNUAL INSPECTION

## (Check whether)

☒ OPERATION LIMITATIONS FORM ACA-309 WAS ISSUED, OR☐ APPROVED AIRPLANE FLIGHT MANUAL IS IN THE AIRCRAFT

## FINDINGS

☒ AIRWORTHY☐ UNAIRWORTHY

DESIGNEE'S SIGNATURE AND NO.

H.E. Wendt, DAMI 9741

CAA INSPECTOR'S SIGNATURE

☐ ACCEPTED☐ REINSPECTED

DATE

July 16, 1949

DATE

REASON FOR DISAPPROVAL, OR REMARKS (Indicate if you have used the reverse to continue this or  
other item ☐ Yes ☐ No)



# WEIGHT AND BALANCE REPORT.

NC 1048V

Make Grumman.

Model G-21A

Serial # 37793

	<u>SCALE</u>	<u>NET</u>
Right Wheel	3380	
Left Wheel	3430	
Tail Wheel	450	
	7260	

Datum to Front Wheel C/l 10.25 ins.

From Front Wheel C/l to Tail Wheel C/l 203.25 ins.

Aircraft weighed with 15 gals. Oil and 220 gals. Fuel.

$450 \times 203.25 = 12.59 + 10.25 = 22.84 \text{ ins.}$   
7260

As weighed	7260	+	22.84		+ 165818.4
Less Fuel	1320	+	30		- 39600
Less Oil	112.5	+	8		- 900
Emtpy C.G.	5827.5	+	21.5		+ 125318.4

	<u>Arm.</u>	<u>Forward C.G.</u>		<u>Rearward C.G.</u>	
		<u>Wt.</u>	<u>Moment</u>	<u>Wt.</u>	<u>Moment</u>
Empty	+21.5	5827.5	+ 125318.4	5827.5	+ 125318.4
Oil	+ 8	112.5	+ 900	112.5	+ 900
Fuel	+30	450	+ 13500	450	+ 13500
Pilots	- 5	340	- 1700	170	- 850
Passengers	+24			---	---
" "	+62			340	+ 21080
" "	+102			340	+ 34680
Baggage	+169			325	+ 54925
		6730	+ 138018.4	7565.0	+ 249553.4

Forward C.G. + 20.507      Rearward C.G. + 32.98

C.G. Limits + 20.5 to 33.0

Gross Weight 8000 lbs.

Weighed by B.G. Airways Ltd.

Witnessed by H.E. Wendt, 9741, DAMI.



## REPAIR AND ALTERATION FORM (AIRCRAFT, PROPELLERS, ENGINES, INSTRUMENTS)

INSTRUCTIONS - This form must be filled out in duplicate each time a major repair and/or alteration is made of an aircraft, propeller, engine, or instrument, as follows:

- (A) For an Aircraft - Complete items 1, 2, 3a, 4, 5, 6, and 7 and submit to CAA representative for approval.  
(B) For a Component Installed in an Aircraft - Complete items 1, 2, 3(b, c, or d, whichever is applicable), 4, 5, and 6, and submit as described in (A) above.  
(C) For a Spare Component - Complete items 3(b, c, or d), 5, and 6, and submit to CAA representative for approval. When approved, retain both copies of this form with the component until installation on an aircraft. At that time, items 1, 2, and 4 must be completed by the installing agency, which will then forward forms as described in (A) above.

1. AIRCRAFT	MAKE GRUMMAN	MODEL G21A	SERIAL NO. 37793	CAA IDENTIFICATION MARK NC1048V
2. OWNER	NAME (First, middle, last) Williams, Arthur James			
	ADDRESS (Street and number, city, zone, and state) 32 Main & Hope Streets, Georgetown, Br. Guiana			

## 3. FILL IN INFORMATION IN THIS ITEM ONLY FOR THE UNIT REPAIRED AND/OR ALTERED

UNIT	MAKE	MODEL	SERIAL NO.	NATURE OF WORK (Check)	
				MAJOR REPAIR	MAJOR ALTERATION
a. AIRCRAFT	(As described in item 1 above)				
b. PROPELLER BLADE OR HUB					
c. ENGINE					
d. INSTRUMENT	TYPE AND MANUFACTURER				

The following items are to be completed by repair or alteration agency. However, in the case of a spare component, item 4 will not be completed until such component is installed in an aircraft. At this time, item 4 will be completed by the installing agency, if applicable.

4. AIRCRAFT	EMPTY WEIGHT (Pounds)* 5827.5	EMPTY CENTER OF GRAVITY (Inches from datum)* 21.5	USEFUL LOAD (Pounds)* 2172.5
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\*AFTER the repairs and/or alterations described below were made.

5. KIND OF AGENCY WHICH MADE REPAIRS AND/OR ALTERATIONS (Check)	
<input type="checkbox"/> MANUFACTURER	<input type="checkbox"/> APPROVED REPAIR STATION NO. _____
	<input checked="" type="checkbox"/> CERTIFIED MECHANIC
	(SPECIFY)

6. AGENCY	NAME B.G. Airways Ltd.	ADDRESS (Street and number, city, zone, and state) 32 Main & Hope Sts., Georgetown British Guiana	DATE WORK ACCOMPLISHED July 15, 1949
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DESCRIPTION OF WORK ACCOMPLISHED IN ACCORDANCE WITH PART 18 OF THE CIVIL AIR REGULATIONS AND ITS ASSOCIATED CIVIL AERONAUTICS MANUAL 18. (If more space is needed, continue on reverse, or attach separate sheets bearing aircraft identification mark)

Ship completely overhauled. All fabric covered surfaces recovered with Grade A Fabric. Tail section replaced from Station #28 to #36 with Grumman assembly. P.&W. R985 AN3 engines #12415 and #16664, overhauled by approved repair station #3604, installed. All work accomplished according to CAM 18. Ship test flown.

I CERTIFY THAT THE ABOVE STATEMENTS ARE TRUE AND CORRECT TO THE BEST OF MY KNOWLEDGE.

James Bow

SIGNATURE OF SUPERVISING MECHANIC

A & E 852

CERTIFICATE NUMBER AND RATING

July 15, 1949

DATE

## TO BE COMPLETED BY CAA REPRESENTATIVES

<input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED <input type="checkbox"/> FORWARDED FOR ENGINEERING APPROVAL	SIGNATURE OF DESIGNEE H.E. Wendt	NUMBER DAMI 9741	DATE July 16, 1949
	SIGNATURE OF INSPECTOR J.E. Wendt	<input type="checkbox"/> ACCEPTED <input type="checkbox"/> REINSPECTED	DATE

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## REPAIR AND ALTERATION FORM (AIRCRAFT, PROPELLERS, ENGINES, INSTRUMENTS)

INSTRUCTIONS - This form must be filled out in duplicate each time a major repair and/or alteration is made of an aircraft, propeller, engine, or instrument, as follows:

- (A) For an Aircraft - Complete items 1, 2, 3a, 4, 5, 6, and 7 and submit to CAA representative for approval.  
(B) For a Component Installed in an Aircraft - Complete items 1, 2, 3(b, c, or d, whichever is applicable), 4, 5, and 6, and submit as described in (A) above.  
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1. AIRCRAFT	MAKE	MODEL	SERIAL NO.	CAA IDENTIFICATION MARK
2. OWNER	NAME (First, middle, last)		ADDRESS (Street and number, city, zone, and state)	
	AIR CARRIER ENGINE SERVICE, Inc.		International Airport, Miami, Florida	

## 3. FILL IN INFORMATION IN THIS ITEM ONLY FOR THE UNIT REPAIRED AND/OR ALTERED

UNIT	MAKE	MODEL	SERIAL NO.	NATURE OF WORK (Check)	
				MAJOR REPAIR	MAJOR ALTERATION
a. AIRCRAFT	(As described in item 1 above)				
b. PROPELLER MADE OR HUB					
c. ENGINE	Pratt Whitney	R-985 AN3	12415	x	
d. INSTRUMENT	TYPE AND MANUFACTURER				

The following items are to be completed by repair or alteration agency. However, in the case of a spare component, item 4 will not be completed until such component is installed in an aircraft. At this time, item 4 will be completed by the installing agency, if applicable.

4. AIRCRAFT	EMPTY WEIGHT (Pounds)*	EMPTY CENTER OF GRAVITY (Inches from datum)*	USEFUL LOAD (Pounds)*
-------------	------------------------	--	-----------------------

\*AFTER the repairs and/or alterations described below were made.

5. KIND OF AGENCY WHICH MADE REPAIRS AND/OR ALTERATIONS (Check) 3604	
<input type="checkbox"/> MANUFACTURER	<input checked="" type="checkbox"/> APPROVED REPAIR STATION NO. (SPECIFY)
	<input type="checkbox"/> CERTIFIED MECHANIC

6. AGENCY	NAME	ADDRESS (Street and number, city, zone, and state)	DATE WORK ACCOMPLISHED
	AIR CARRIER ENGINE SERVICE, Inc.	International Airport, Miami, Fla.	9-5-47

7. DESCRIPTION OF WORK ACCOMPLISHED IN ACCORDANCE WITH PART 18 OF THE CIVIL AIR REGULATIONS AND ITS ASSOCIATED CIVIL AERONAUTICS MANUAL 18. (If more space is needed, continue on reverse, or attach separate sheets bearing aircraft identification mark)

Engine given major overhaul - completely disassembled, cleaned, inspected and all steel parts magnafluxed one hundred percent. Replacements made where necessary with following major replacements made: two 31795 piston pins, one 20416 bearing, 5max four 12768 bearings, two RT52137 rotor, one SP5287 cam ~~III~~ follower, three BB5210 bearings, one BB525 bearing, two LR528 cam lub., one 32983 bearing. Engine reassembled and sent with overhauled magnetos and carburetor to be block tested according to factory specification. Test satisfactory. All applicable service bulletins incorporated.

I CERTIFY THAT THE ABOVE STATEMENTS ARE TRUE AND CORRECT TO THE BEST OF MY KNOWLEDGE.

SIGNATURE OF SUPERVISING MECHANIC	CERTIFICATE NUMBER AND RATING	DATE
William C. Bergeton	M15353	9-30-47

## TO BE COMPLETED BY CAA REPRESENTATIVES

<input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED <input type="checkbox"/> FORWARDED FOR ENGINEERING APPROVAL	SIGNATURE OF DESIGNEE	NUMBER	DATE
	SIGNATURE OF INSPECTOR	<input type="checkbox"/> ACCEPTED <input type="checkbox"/> REINSPECTED	DATE



## REPAIR AND ALTERATION FORM (AIRCRAFT, PROPELLERS, ENGINES, INSTRUMENTS)

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- (A) For an Aircraft - Complete items 1, 2, 3a, 4, 5, 6, and 7 and submit to CAA representative for approval.  
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(C) For a Spare Component - Complete items 3(b, c, or d), 5, and 6, and submit to CAA representative for approval. When approved, retain both copies of this form with the component until installation on an aircraft. At that time, items 1, 2, and 4 must be completed by the installing agency, which will then forward forms as described in (A) above.

1. AIRCRAFT	MAKE	MODEL	SERIAL NO.	CAA IDENTIFICATION MARK
2. OWNER	NAME (First, middle, last)		ADDRESS (Street and number, city, zone, and state)	
	AIR CARRIER ENGINE SERVICE, Inc.		International Airport, Miami, Florida	

## 3. FILL IN INFORMATION IN THIS ITEM ONLY FOR THE UNIT REPAIRED AND/OR ALTERED

UNIT	MAKE	MODEL	SERIAL NO.	NATURE OF WORK (Check)	
				MAJOR REPAIR	MAJOR ALTERATION
a. AIRCRAFT	(As described in item 1 above)				
b. PROPELLER BLADE OR HUB					
c. ENGINE	Pratt Whitney	R-985 AN3	16664	x	
d. INSTRUMENT	TYPE AND MANUFACTURER				

The following items are to be completed by repair or alteration agency. However, in the case of a spare component, item 4 will not be completed until such component is installed in an aircraft. At this time, item 4 will be completed by the installing agency, if applicable.

4. AIRCRAFT	EMPTY WEIGHT (Pounds)*	EMPTY CENTER OF GRAVITY (Inches from datum)*	USEFUL LOAD (Pounds)*
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\*AFTER the repairs and/or alterations described below were made.

5. KIND OF AGENCY WHICH MADE REPAIRS AND/OR ALTERATIONS (Check)	
<input type="checkbox"/> MANUFACTURER	<input checked="" type="checkbox"/> APPROVED REPAIR STATION NO. 3604
	<input type="checkbox"/> CERTIFIED MECHANIC
(SPECIFY)	

6. AGENCY	NAME	ADDRESS (Street and number, city, zone, and state)	DATE WORK ACCOMPLISHED
	AIR CARRIER ENGINE SERVICE, Inc.	International Airport, Miami, Fla.	9-5-47

## 7. DESCRIPTION OF WORK ACCOMPLISHED IN ACCORDANCE WITH PART 18 OF THE CIVIL AIR REGULATIONS AND ITS ASSOCIATED CIVIL AERONAUTICS MANUAL 18. (If more space is needed, continue on reverse, or attach separate sheets bearing aircraft identification mark)

Engine given major overhaul - completely disassembled, cleaned, inspected and all steel parts magnafluxed one hundred percent. Replacements were made where necessary with following major replacements made: four 12678 bearing, two 109 bearing, one 20416 bearing, one 8797 gear, one 5017 race, six 31795 piston pins, one 32983 bearing. Engine reassembled and sent with newly overhauled magnetos and carburetor to be block-tested according to factory specification. Test satisfactory. All applicable service bulletins were incorporated.

I CERTIFY THAT THE ABOVE STATEMENTS ARE TRUE AND CORRECT TO THE BEST OF MY KNOWLEDGE.

SIGNATURE OF SUPERVISING MECHANIC	CERTIFICATE NUMBER AND RATING	DATE
William C. Bergstein	E-1 15353	9-30-47

## TO BE COMPLETED BY CAA REPRESENTATIVES

<input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED <input type="checkbox"/> FORWARDED FOR ENGINEERING APPROVAL	SIGNATURE OF DESIGNEE	NUMBER	DATE
	SIGNATURE OF INSPECTOR	<input type="checkbox"/> ACCEPTED <input type="checkbox"/> REINSPECTED	DATE



# LIST OF EQUIPMENT.

Item No.	Class	Name of Item	Arm	Weight
201		Constant speed propeller controls	-23	13
302		Two controllable metal propellers		
		Ham.Std, hubs 2D30-237		
		Blades 6101A-18	-52	308
101		Two engine ring cowls	-36	56
102		Two oil radiators(UAP #U3160)	0	26
103		Two starters(Eclipse E-160)	-12	77
108		Pressure Fire extinguisher(Lux type (Lux type A-1)	-33	21
111		Engine		684
218		Engine tool kit(stowed)	+161	
221		Engine work shelf "	+189	
106		30x7 wheels 32x8 ply tyres (G3-56A Hayes wheels)		
301		Wing Tip Floats		
305		Tail wheel 12.50 smooth contour with Goodrich 6 ply special heavy duty tyre(wheel must be placarded for this type tyre)		
104		Generator 50 amp.(Eclipse)	-23	31
105		Batteries Two.38A(Exide 6-TS-9-L)	-7	75
203		Extra generator 50 amp. (Eclipse ")	-23	31
215		One landing light(Grimes ST-1000)	+60	8
207		Learadio receiver RCBB,tuning unit	-32	7
		Lear transmitter UT6A, and receiver power supply	+151	45
205		Cabin furnishings		
		(a) Leather chair upholstery	+77	40
		(b) Four sets quick detachable chair leg locks.	+77	12
217		Miscellaneous seaplane equipment.		
		(a) Anchor (Northill) 12 lb & 150 ft Nylon line 11/16 dia.	-88	34
		(b) Bilge pump,boat hook, and fog horn.		7

DEPT OF COMMERCE  
AERONAUTICS ADM

OCT 24 1 06 PM '48

MAIL ROOM 3  
WASHINGTON

UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

M. W. Leaphart

(Inspector's signature)

Page 1 of 6 pages

Date May 22, 1945

AIRCRAFT OPERATION RECORD

(Non Air Carrier)

Grumman Aircraft Engineering Corp.

(JRF-5)

37793

NR 46497

7 PCAMFbM (Manufacturer)

Feb. 1944 (Model)

(Serial number)

(CAA Identification mark)

PLL-AM-FbM

(Designation)

(Date manufactured)

(Aircraft specification number)

(Incidence)

Wing leading edge at wing butt.

(M. A. C.)

(Datum to L. E. of M. A. C.)

Datum used

Leveling lugs in pilots compartment

Leveling means

SEATING ARRANGEMENT

Number	Location or arm	Number	Location or arm	Number	Location or arm
2	-5"				
1	+24"				
2	+62"				
2	+102"				

BAGGAGE COMPARTMENTS

Capacity	Location or arm	Capacity	Location or arm
300 lb.	in bow compartment -65"		
105 lb.	in rear compartment +169"		

FUEL TANKS

280 gals.	110 gal. in each wing at +30"
	60 gal. in aux. tank in fuselage +27"

OIL TANKS

15 gals.	7 1/2 gal. in each nacelle +8"
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OPERATIONS AUTHORIZED

1. INSTRUMENT NIGHT (not for hire at night)
2. OPERATE FROM SHELTERED WATER OR PREPARED LANDING AREAS AS PER NAVY ORDER 96-42
3. PASSENGERS AUTHORIZED ARE: Representatives of the Rubber Development Corporation and/or individuals flying in the interest of the Rubber Development Corporation.  
No passenger shall accompany ferry flight from Miami to Manaus, Brazil.
4. When two crew in pilots compartment, any amount of fuel, not more than 25 lbs. is to be carried in BOW BAGGAGE COMPARTMENT.
5. With SEVEN PERSONS aboard, 405 lbs. BAGGAGE, ONLY 132 Gals. FUEL PERMITTED.
6. With FULL FUEL?, NO BAGGAGE, FOUR PERSONS ONLY MAY BE CARRIED.
7. Maintain loading conditions of aircraft so that Center of Gravity Limits are not exceeded in any Gross loaded weight.

## NOTES

If Operation Record form, or any part thereof, is lost or destroyed, a duplicate may be obtained from the Civil Aeronautics Administration, Washington, D. C., at the rate of \$1 per page. (Money order or check should be made payable to the Treasurer of the United States.)

When, following examination and inspection of the aircraft to which this form is issued, endorsement of the **Aircraft Airworthiness Certificate** is refused, a copy of the "disapproved" **Aircraft Inspection Report** will be attached hereto, and will remain herewith until removed by an inspector of the Administration following a satisfactory inspection of the aircraft.

To execute those portions of this form pertaining to, "**Datum Used**," "**C. G. Limits**," and "**Leveling Means**," use the figures indicated in the pertinent aircraft specifications, where such information is given; otherwise, when established, describe.

Before adding, removing, or relocating any item of equipment on the aircraft to which this form is issued, or making any repair or alteration, consult Part 18 of the Civil Air Regulations, or an inspector of the Administration.

When additional space is required for the listing of "**Equipment**" and "**Major Repairs**," as provided for on pages 3 and 5, respectively, of this form, additional sheets similar to those to which reference is made, will be inserted and made a part of this form by an inspector of the Administration.

Sheets issued as a part of this form shall not be added to, removed from, or altered in any manner, except by an inspector of the Administration.



UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

*J. Allen McPeake*  
J. Allen McPeake  
(Inspector's signature)

Page 1 of 6 pages

Date May 16, 1945

AIRCRAFT OPERATION RECORD  
(Non Air Carrier)

Grumman Aircraft Engineering G-21-A (JRF-5) 37793 NR 46497  
(Manufacturer) (Model) (Serial number) (CAA identification mark)  
7 PCAmFbM Feb. 1944 654  
(Designation) (Date manufactured) (Aircraft specification number)

(Incidence) (Dihedral) (Stagger) (M. A. C.) (Datum to L. E. of M. A. C.)  
Datum used Wing leading edge at wing butt.  
(Obtain from pertinent aircraft specifications or give complete description)

Leveling means Leveling lugs in pilots compartment

## SEATING ARRANGEMENT

Number	Location or arm	Number	Location or arm	Number	Location or arm
2	-5"				
1	24"				
2	62"				
2	102"				

## BAGGAGE COMPARTMENTS

Capacity	Location or arm	Capacity	Location or arm
300 lb.	in bow compartment -65"		
105 lb.	in rear compartment. 169"		

## FUEL TANKS

280 gals. 110 gal in each wing at 30"  
60 gal. in aux. tank in fuselage 27"

## OIL TANKS

15 gals. 7½ gal in each nacelle 8"

## OPERATIONS AUTHORIZED

1. INSTRUMENT NIGHT (not for hire at night)
2. OPERATE FROM SHELTERED WATER OR PREPARED LANDING AREAS AS PER NAVY ORDER 96-42.
3. USE ONLY FOR CARGO AND/OR CARRIAGE OF EMPLOYEES OF RUBBER DEVELOPMENT CORP. AND CREW ESSENTIAL TO PURPOSES OF FLIGHT.
4. When two crew in pilots compartment, any amount of fuel, not more than 25 lb. is to be carried in BOW BAGGAGE COMPARTMENT.
5. With SEVEN PERSONS aboard, 405 pounds BAGGAGE, ONLY 132 Gals. FUEL PERMITTED.
6. With FULL FUEL?, NO BAGGAGE, FOUR PERSONS ONLY MAY BE CARRIED.
7. Maintain loading conditions of aircraft so that Center of Gravity Limits are not exceeded in any Gross loading weight.

## NOTES

If Operation Record form, or any part thereof, is lost or destroyed, a duplicate may be obtained from the Civil Aeronautics Administration, Washington, D. C., at the rate of \$1 per page. (Money order or check should be made payable to the Treasurer of the United States.)

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To execute those portions of this form pertaining to, **"Datum Used," "C. G. Limits,"** and **"Leveling Means,"** use the figures indicated in the pertinent aircraft specifications, where such information is given; otherwise, when established, describe.

Before adding, removing, or relocating any item of equipment on the aircraft to which this form is issued, or making any repair or alteration, consult Part 18 of the Civil Air Regulations, or an inspector of the Administration.

When additional space is required for the listing of **"Equipment"** and **"Major Repairs,"** as provided for on pages 3 and 5, respectively, of this form, additional sheets similar to those to which reference is made, will be inserted and made a part of this form by an inspector of the Administration.

Sheets issued as a part of this form shall not be added to, removed from, or altered in any manner, except by an inspector of the Administration.

UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

J. Allen McPeck  
(Inspector's signature)

Page 1 of 6 pages

Date May 16, 1945

AIRCRAFT OPERATION RECORD  
(Non Air Carrier)

Grumman Aircraft Engineering

(Manufacturer)

G-21-A (JRF-5)

(Model)

37793

(Serial number)

NR 46497

(CAA identification mark)

7 PCamFbm

(Designation)

Feb. 1944

(Date manufactured)

654

(Aircraft specification number)

(Incidence)

(Dihedral)

(Stagger)

(M. A. C.)

(Datum to L. E. of M. A. C.)

Datum used Wing leading edge at wing butt.

(Obtain from pertinent aircraft specifications or give complete description)

Leveling means Leveling lugs in pilots compartment.

SEATING ARRANGEMENT

Number	Location or arm	Number	Location or arm	Number	Location or arm
--------	-----------------	--------	-----------------	--------	-----------------

2

5"

1

24"

2

62"

2

102"

BAGGAGE COMPARTMENTS

Capacity	Location or arm	Capacity	Location or arm
----------	-----------------	----------	-----------------

300 lb. in bow compartment -65"

105 lb. in rear compartment. 169"

FUEL TANKS

280 gals. 110 gal in each wing at 30"

60 gal. in aux. tank in fuselage 27"

OIL TANKS

15 gals. 7 1/2 gal in each nacelle 8"

OPERATIONS AUTHORIZED

1. INSTRUMENT NIGHT (not for hire at night)
2. OPERATE FROM SHELTERED WATER OR PREPARED LANDING AREAS AS PER NAVY ORDER 96-42.
3. USE ONLY FOR CARGO AND/OR CARRIAGE OF EMPLOYEES OF RUBBER DEVELOPMENT CORP. AND CREW ESSENTIAL TO PURPOSES OF FLIGHT.
4. When two crew in pilots compartment, any amount of fuel, not more than 25 lb. is to be carried in BOW BAGGAGE COMPARTMENT.
5. With SEVEN PERSONS aboard, 405 pounds BAGGAGE, ONLY 132 Gals. FUEL PERMITTED.
6. With FULL FUEL?, NO BAGGAGE, FOUR PERSONS ONLY MAY BE CARRIED.
7. Maintain loading conditions of aircraft so that Center of Gravity Limits are not exceeded in any Gross loaded weight.

### NOTES

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To execute those portions of this form pertaining to, "Datum Used," "C. G. Limits," and "Leveling Means," use the figures indicated in the pertinent aircraft specifications, where such information is given; otherwise, when established, describe.

Before adding, removing, or relocating any item of equipment on the aircraft to which this form is issued, or making any repair or alteration, consult Part 18 of the Civil Air Regulations, or an inspector of the Administration.

When additional space is required for the listing of "Equipment" and "Major Repairs," as provided for on pages 3 and 5, respectively, of this form, additional sheets similar to those to which reference is made, will be inserted and made a part of this form by an inspector of the Administration.

Sheets issued as a part of this form shall not be added to, removed from, or altered in any manner, except by an inspector of the Administration.

*J. Allen McPeck*  
(Inspector's signature)

46497

(Identification mark)

Date May 16, 1945

## OPERATION LIMITATIONS

### 1. AIRSPEEDS—NEVER TO BE EXCEEDED

Level flight or climb 194 mph ~~169 knots~~ indicated.

Glide or dive 225 mph ~~195 knots~~ indicated.

Flaps extended 110 mph ~~96 knots~~ indicated.

NOTE.—These airspeed limitations are based on values for which the structure has been designed or tested, and are maximum values which shall not be exceeded. THEY ARE NOT RECOMMENDED OPERATION AIRSPEEDS.

### 2. ENGINE

#### TAKE-OFF (1 MINUTE)

At any altitude 36.5 in. hg. 2300 r. p. m. (450 hp.)

#### MAXIMUM, EXCEPT TAKE-OFF

At sea level 34.5 33.5 in. hg. 2200 r. p. m. (380 <sup>400</sup> hp.)

At 5000 feet altitude 33.5 in. hg. 2200 r. p. m. (400 hp.)

(For additional limitations see opposite side of this sheet)

NOTE.—Manifold pressure limit varies linearly between sea level and altitude designated.

These limits apply only with the power plant described below:

Engine manufacturer	Model	Number installed
<u>Jacobs Aircraft Engine Co.</u>	<u>Wasp R-985-AN-15</u>	<u>2</u>
	<u>an-6B</u>	
	<u>similar to SB-3.</u>	

Propeller manufacturer	Hubs	Blades	Number installed
<u>Hamilton Standard Propellers</u>	<u>2D30</u>		<u>2</u>
<u>Div. of. United Aircraft Corp.</u>		<u>6167A-12</u>	<u>4</u>
	<u>(constant speed)</u>		

### 3. WEIGHT

The maximum certificated weight of this aircraft which shall not be exceeded, is:

8800 (Land, with wheels) 8800 (Water type)

### 4. BALANCE

This aircraft shall be so loaded at all times that the center of gravity falls within the following limits:

Center of gravity limits—Land type plus 20.5" to 33.0"

Center of gravity limits—Water type plus 20.5" to 33.0"

(OVER)

10

## ADDITIONAL OPERATION LIMITATIONS

### REMARKS

NOTE.—Following any modification to the aircraft to which this form is issued or to the pertinent aircraft specifications which necessitates a change in the information contained on this sheet, a superseding sheet will be issued by an inspector of the Authority, and made a part hereof. The superseded sheet with a copy of the superseding sheet will be forwarded with the pertinent file.

UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

J. Allen MoPeeke

(Inspector's signature)

46497

(Identification mark)

Date May 16, 1945

## EQUIPMENT

Item No. (1)	Class (2)	Name of Item (3)	Arm (4)	Weight (5)
101 ✓		Two engine ring cowl	- 36"	56 lb.
102 ✓		Two oil radiators UAP-U3160	0	26
103 ✓		Two starters Eclipse E-160	- 12	77
203(c) 104 ✓		Generator 50 amp. Eclipse E-5	- 23	26
203(c) 105 ✓		Generator 50 Amp. Eclipse E-5 (Extra Gen.)	- 23	17*
106 ✓		Two batteries Exide 6TS-13-1	- 7	75
Special 107 ✓		30x7 Wheels (Hayes G-3-56-A) 32-8 tires 8 ply.	-	-
107 ✓		13.25 tail wheel with 6-ply heavy duty tire	-	-
108 ✓		Lux type pressure fire extinguisher A-1	- 33	21
202 302 ✓		Two controllable Ham. Std. Propellers 2D30 With 6167A-12 blades.	- 52	308
301 ✓		Wind tip floats - Grumman	-	87
201 ✓		Constant speed propeller controls	- 23	13
222 ✓		Aux. Fuel tank (80 gal.) Grumman 12900 C	27	40
207 ✓		Radio installation	- 60	150
		Bendix Receiver R411		
		Bendix Transmitter T421		
		12 V. Dynamo		
		Control Box for transmitter		
		Remote control for Receiver		
215 ✓		One Griggs ST 1000 Landing light (Left)	68	8
Special Reql		Two pilot chairs and cushions and safety belts.		
"		Four passenger chairs and belts		
"		One small passenger seat and belt		
"		Two R&W Wasp engines model R895-AK-63		
"		Navigation lights		
"		Primary Flight Instruments		
"		Seven Inflatable type life belts.		
"		Three Jungle kits.		
"		Two shot-guns and shells.		
"		One Very's Pistol and shells.		
"		One Machetti.		
"		One box (25 lb.) K rations.		
"		Mechanics tools and parts kit.		
"		Toilet.		
"		One Gallon fresh water container		
		J. Allen MoPeeke		

## NOTES

*Column (1).*—Use item number listed on pertinent aircraft specification. If old specifications and no number is listed, leave this space blank. If item is not included on pertinent aircraft specification, mark "Special."

*Column (2).*—Designate whether Class I, II, or III. Class I denotes equipment which must be installed unless replaced by Class II or III equivalent items, or unless designated for "balance" only. Equivalent Class II or III items should then be marked "Required" (as, II Req." or "III Req."). Class II denotes items which may be installed or removed "subject to inspection and check of balance." Class III denotes equipment which may be installed or removed subject to inspection only, except when some item of Class II equipment is installed, in which case check of balance is also required. Reference above to Class III equipment does not apply to those aircraft on which specifications require weight and balance report for original certification, and following any change in equipment thereafter.

*Column (3).*—Include name of manufacturer, type, and/or model when necessary to complete identification of item.

*Column (4).*—Use Arm listed on Pertinent Aircraft Specification. If no Arm or Datum is listed on pertinent specifications, **specify datum** used, when arm is established.

For those aircraft previously certificated, on which equipment has not been classified (as Classes I, II, and III), Columns (1), (2), and (4) may be left blank. The location of equipment subsequently added, removed or relocated, will be clearly described in that portion of this form dealing with "Weights and Alterations," and included on the Repair and Alteration Form covering the alteration, which is submitted.

When any equipment is added, removed, or relocated, the sheet thus affected will be superseded by new "Equipment" sheet, and the old sheet together with a copy of the superseding sheet will be forwarded with the pertinent file.

Inspectors signature will appear immediately below last item of equipment listed, which was installed at the time inspected and approved by him, if equipment listed has not previously been approved and endorsed by an inspector of the Administration.



(Identification mark)

J. Allen McPeeke

(Inspector's signature)

Date May 16, 1945

## WEIGHTS AND ALTERATIONS

WEIGHTS	Land	Water	Ski
Empty (actual)	6309 / lbs.	lbs.	lbs.
Useful *	2491 / lbs.	lbs.	lbs.
Gross *	8800 / lbs.	lbs.	lbs.

See Loading Schedule dated see page one, where such is required, and any revision thereto, which is made a part hereof.Empty center of gravity is 21.92 inches—forward ( ) aft (XX)—of Datum

## ALTERATION

Alteration of \_\_\_\_\_ checked by \_\_\_\_\_  
(Date) (Inspector)

described as follows: \_\_\_\_\_

accomplished by \_\_\_\_\_  
(Name and number)for which copy of Repair and Alteration Form dated \_\_\_\_\_ was submitted.  
The following special notes listed on the pertinent aircraft specifications have been complied with on the dates shown:

Special note	Date	Special note	Date	Special note	Date
No. <u>AD-317</u> <u>None</u> <u>Applicable</u>		No. _____		No. _____	
No. <u>SN-1+2</u> <u>not</u>		No. _____		No. _____	
No. _____		No. _____		No. _____	
No. _____		No. _____		No. _____	

REMARKS: \_\_\_\_\_

This page superseded \_\_\_\_\_  
(Date)

\* Weights which are not to be exceeded.

(Inspector's signature)

(See notes on reverse side of this sheet, pertaining to information provided for.)

### NOTES

*The following values shall be used when computing useful loads:*

Fuel at 6 pounds per gallon, lubricating oil 7.5 pounds.

All occupants, baggage, and other useful load items at actual weight.

---

The information provided for hereon will be maintained in a current condition.

Following any alteration affecting any information contained hereon, a superseding sheet will be issued by an inspector of the Administration, and made a part hereof. The duplicate copy of the superseding sheet will be forwarded with the pertinent file.

All sheets pertaining to repairs and alterations, which have been superseded, will be retained as a part of this form for record purposes.

RECEIVED  
CERTIFICATE SECTION  
MAY 22 12 01 PM '45

CLASSIFICATION	NATURE OF WORK PERFORMED	
	Major repair	Major alteration
AIRCRAFT	<b>X</b>	
ENGINE		
PROPELLER		
INSTRUMENT		

UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

(To be filled in by inspector)	
AFTER ACCIDENT OF	(Date)
APPROVED	
DISAPPROVED	
Forwarded for engineering approval (Submit complete details)	
Date	
Inspector	

*Engineering OK  
A-290 act. 9/19/45  
B. L. Springer*

*File  
PC 14  
nm*

**REPAIR AND ALTERATION FORM**  
**AIRCRAFT—ENGINES—PROPELLERS—INSTRUMENTS**

Aircraft Identification Mark No. **NR-46497**

Owner's name Rubber Development Corporation, Manaus Office, Brazil,  
 Owner's address HOLC Bldg., 101 Indiana Ave. NW., Washington 25, D. C.  
 Aircraft manufacturer and model Grumman - G-21-A Serial No. 37793  
 Engine manufacturer and model\* \_\_\_\_\_ Serial No. \_\_\_\_\_  
 Propeller manufacturer and model\* \_\_\_\_\_ Serial No. \_\_\_\_\_  
 Propeller blade model\* \_\_\_\_\_ Serial No. \_\_\_\_\_  
 Propeller hub model\* \_\_\_\_\_ Serial No. \_\_\_\_\_

\*To be filled in only for unit repaired or altered.

**REPAIR OR ALTERATION AGENCY**

Manufacturer. \_\_\_\_\_ Approved repair station No. \_\_\_\_\_ **X** Certificated mechanic  
 Agency's name Rubber Development Corporation  
 Address Manaus, Brazil Date of repair or alteration Aug. 17, 1945

The following work has been accomplished in accordance with Part 18, Civil Air Regulations. (For recommended practice refer to CAM 18.)

[If blank space is insufficient for clear presentation, attach separate pages (8" x 10½" or multiples thereof) bearing aircraft identification mark]

**36" of Chine replaced on left side of hull from wheel well 36" forward.**

**Replacement made with factory chine.**

**NOTE: Repair necessary due collision with fence post.**

*2*

MAIL ROOM

FWD

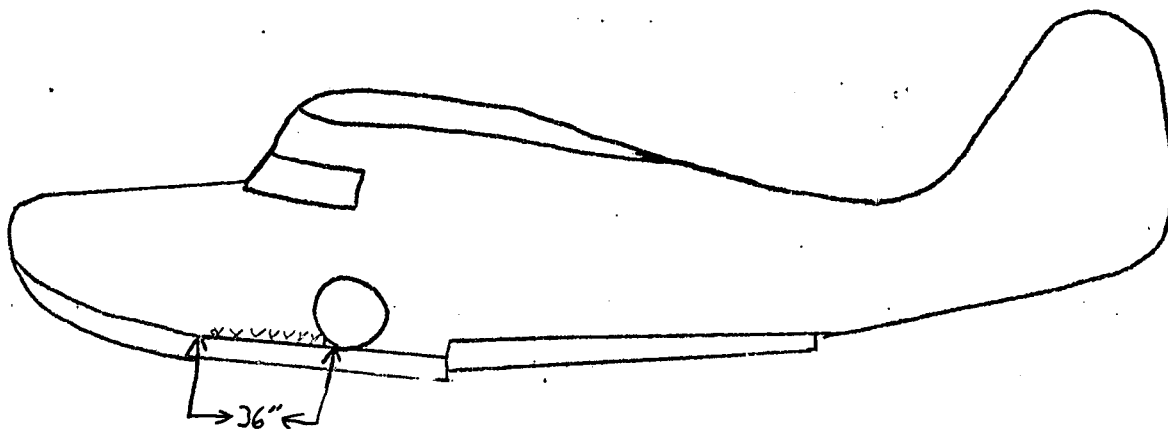
AFT

WHEEL  
WELL

SPICE OF 24ST AND  
.125 THICKNESS

36" FACTORY CHINE

36"



### SUPERVISING MECHANIC'S CERTIFICATE

Hans V. Koehler

certifies that the foregoing statements are true of his own knowledge.

August 17, 1945

(Date)

Hans V. Koehler

(Mechanic's Signature)

Mechanic's Certificate No.

27590

(Rating)

A Y F

DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

STANDARD REFERENCE SLIP

DATE 5/22/45  
To Mr. Morgan Ref. No. A285  
FROM Mr. Tappan Ref. No. 1000  
REMARKS:

*As per our file from  
Mr. Tappan*

*Revised page 1 of  
form 309 dated  
5-22-45,  
transmitted with  
this form ACA 285  
nm*

Check more than one caption where necessary

☐ For your information. ☐ Note. ☐ Return.

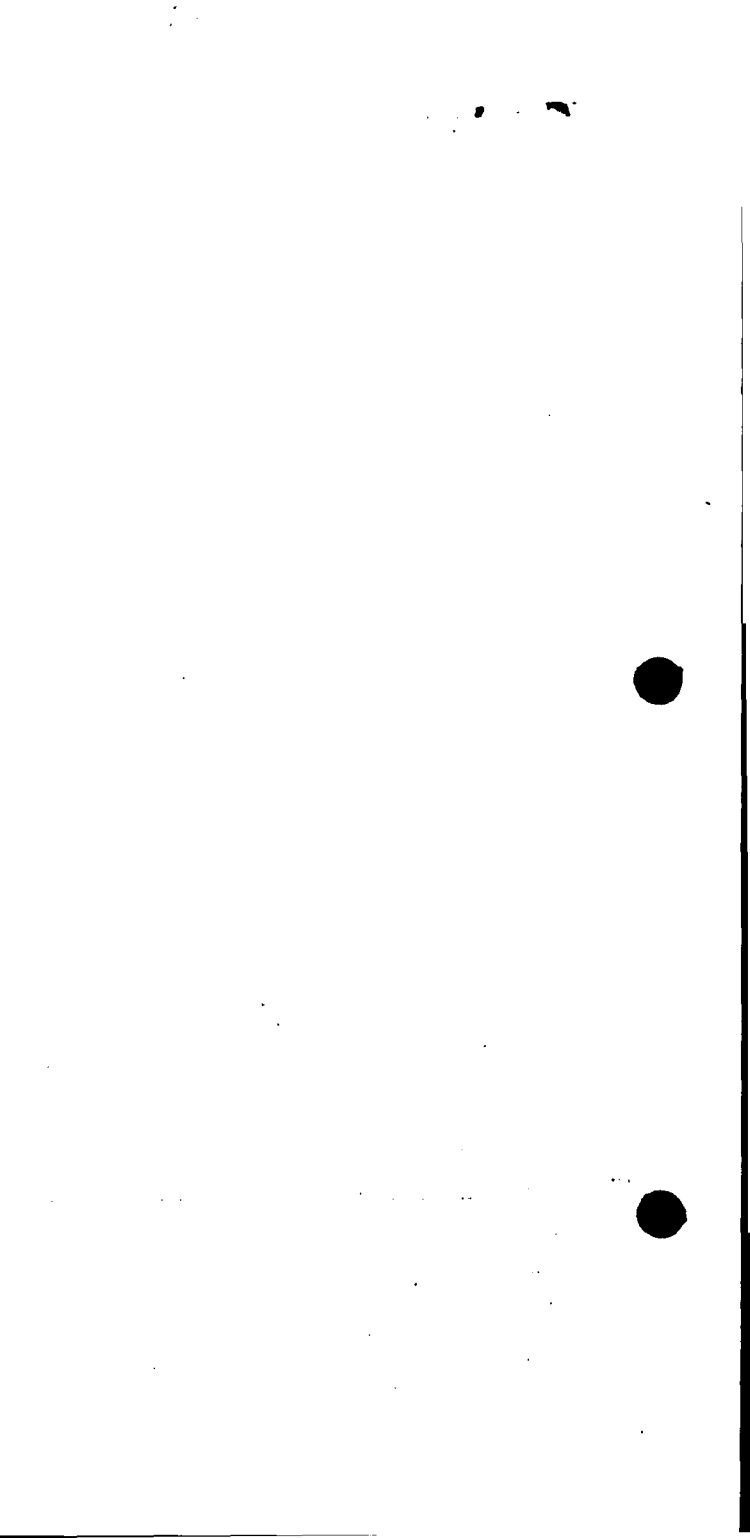
Prepare reply for—

☐ My signature. ☐ Your signature.

Signature of \_\_\_\_\_

Handle—

- ☐ Without recourse. ☐ Comment.  
☐ Investigate and report. ☐ See me.  
☐ Send to central files. ☐ Telephone me.  
☐ Returning file and copies of any correspondence  
to my office.  
☐ Sending only copies of correspondence to my  
office.



FOREIGN FLIGHT AUTHORIZATION

Amendment No. 1

Issued to:

Date issued MAY 19 1945

Rubber Development Corporation  
(A United States Agency)  
101 Indiana Avenue, N.W.  
Washington, D. C.

Reference is made to the foreign flight authorization issued to Rubber Development Corporation (a United States agency), this date, to cover the flight of Grumman aircraft, serial number 37793, identification mark NR46497, from Miami, Florida to Manaus, Brazil, S. A., for operation in Brazil, Colombia, Bolivia and Peru.

Item 4 of the above referred to authorization is hereby amended to read as follows:

4. Passengers. Passengers shall be limited to employees of the Rubber Development Corporation. No passenger shall accompany ferry flight from Miami to Manaus, Brazil.

This amendment shall be attached to and made a part of the foreign flight authorization.

John T. Morgan, Chief  
Aircraft Registration and Recordation

FJEnurrows:abm

CANCELLED

6-4-46  
726





FOREIGN FLIGHT AUTHORIZATION

Issued to:

Date issued MAY 19 1945

Rubber Development Corporation  
(A United States Agency)  
101 Indiana Avenue, N. W.  
Washington, D. C.

Authorization is hereby granted for the foreign flight of a United States registered aircraft from Miami, Florida, to Manaus, Brazil, S. A., via Camaguey, Cuba; Port au Prince, Haiti; Dominican Republic; San Juan, Puerto Rico; Antigua, British West Indies; Port of Spain, Trinidad; Georgetown, British Guiana; Cayenne, French Guiana; Belem and Santarem, Brazil, and for subsequent flights between Manaus and various points in Colombia, Bolivia, and Peru, after arrival at Manaus, premised upon the following express terms and conditions:

1. Purpose. Delivery. Aircraft is to be operated in Brazil, Colombia, Bolivia and Peru, in connection with the activities of the Rubber Development Corporation.
2. Aircraft. The aircraft shall be a Grumman, serial number 37793, identification mark NR46497, registered in the name of the Rubber Development Corporation (a government agency), 101 Indiana Avenue, N.W., Washington, D.C. The aircraft shall at all times be maintained in a condition for safe operation and shall have on board currently effective aircraft registration and airworthiness certificates.
3. Crew.

Pilot. Frank L. Sage, a United States citizen, holder of United States commercial pilot certificate No. 37611.

Copilot and mechanic. John W. Ganz, a United States citizen, holder of United States private pilot certificate No. 57223, and aircraft and aircraft engine mechanic certificate No. M-2861.
4. Passengers. Representatives of the Rubber Development Corporation and/or individuals flying in the interest of the Rubber Development Corporation. No passenger shall accompany ferry flight from Miami to Manaus, Brazil.
5. Radio equipment. Two-way radio.
6. Special apparatus. No camera or firearm shall be carried.



7. Cargo. Aircraft spare parts and/or aviation equipment.

8. First aid equipment. A first aid kit shall be carried and shall contain the items listed for the No. 2 kit, as set forth in Form ACA 1008, dated April 15, 1942, "First Aid Equipment for Aircraft Engaging in Foreign Flight," copy attached.

9. Emergency equipment. The aircraft shall carry a rubber life raft, Navy type life preserver for each person on board, Very pistol and signals.

10. Flight shall be made under daylight flying conditions only.

11. Special conditions for flight in foreign countries.

Cuba. Permission for the aircraft to fly over and land in Cuba shall be obtained by the pilot from the local Cuban consul prior to departure from Miami, and flight within Cuba shall at all times conform to any special conditions or restrictions incorporated in the said permission.

Haiti and Dominican Republic. These countries are parties to the Convention on Commercial Aviation adopted at Habana on February 20, 1928, and in accordance with the terms and conditions of the said Convention, no prior authorization is necessary to fly over and land therein.

The pilot shall notify the authorities at the point of entry in each of these countries prior to arrival, giving the approximate date, hour and place of arrival, the identification of the aircraft and the names and citizenship of persons carried therein. With reference to Haiti, such notice shall be sent to Pan American Airways, the operator of the Port au Prince Airport, in order that attendants and clearance officials may be arranged.

Antigua, Trinidad, British Guiana and French Guiana. The Department of State has advised that permission for the aircraft to fly over and land in Antigua, Trinidad, British Guiana and French Guiana, has been granted by the competent governmental authorities.

Brazil, Colombia, Bolivia and Peru. Flight of the aircraft in these countries shall be in accordance with the air agreements in force between the governments of Brazil, Colombia, Bolivia and Peru, and the United States government.

12. Puerto Rico. No permission is required for flight of the aircraft in Puerto Rico since it is United States territory.

13. General conditions. See item 13 attached.

14. Expiration. Unless sooner suspended or revoked by the Administrator of Civil Aeronautics, this authorization shall continue in effect indefinitely or until termination of service of the aircraft in Brazil, Colombia, Bolivia and Peru, provided that the aircraft and airman certificates are currently in full force and effect during such period.



Rubber Development Corporation - 3

This authorization shall be carried on board the aircraft in the custody of the pilot in command at all times during the flight operation.

John T. Morgan, Chief  
Aircraft Registration and Recordation

Attachments

FJBurrows:abm



Grumman, model G-21A, serial No. 37793, NR46497.

Forms ACA 500-A and ACA 308-A, and ACA 309, issued to Rubber Development Corp. by Inspector McPeeke, on May 16, 1945. Authorized to issue the NR certificate by Mr. Lanter - 8800 lbs. gross wgt. - Mr. Lanter's wire dated May 5, 1945.

Operations authorized.

1. Instrument night, not for hire at night.
2. Operate from sheltered water or prepared landing areas as per Navy Order No. 96-42.
3. Use only for cargo and/or carriage of employees of Rubber Development Corporation and crew essential to purpose of flight.
4. When two crew in pilot compartment any amount of fuel, not more than 25 lbs., is to be carried in bow baggage compartment.
5. With 7 persons aboard, 405/lbs. baggage only, 132 gallons fuel permitted.
6. With full fuel no baggage, 4 persons only may be carried.
7. Maintain loading conditions of aircraft so that center of gravity limits are not exceeded in any gross loaded weight.

Maximum seating - 7

7 PCamb. Flying boat.

73813





UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

THIS CERTIFICATE MUST BE CARRIED IN  
THE AIRCRAFT AT ALL TIMES

*N*  
AIRCRAFT AIRWORTHINESS CERTIFICATE NO. **46497**

This Certifies that **GRUMMAN AIRCRAFT ENGINEERING CORP.**

**G-21-A** *8-21A*

manufacturer's serial No. **37793**, has been inspected and this day found to be in condition for safe operation when operated and maintained in accordance with the regulations and practices prescribed by the Civil Aeronautics Board.

The aircraft for which this certificate is issued (a) shall not be operated unless there is attached hereto the currently effective Aircraft Operation Record issued by the Administrator of Civil Aeronautics for the aircraft, and (b) shall not be operated in flight unless a pilot possessed of a currently effective and appropriate pilot certificate is in command.

This certificate is of 60 days' duration and, unless the holder hereof is otherwise notified within such period, shall continue in effect indefinitely thereafter, unless suspended, revoked, or cancelled by the Civil Aeronautics Board, except that it shall immediately expire (1) at the end of **one year** after the date of issuance of this certificate or after the date of last endorsement hereof, whichever is later, if within such period this aircraft is not examined or inspected by an authorized Civil Aeronautics inspector, or (2) at any time an authorized Civil Aeronautics inspector shall refuse to endorse this certificate after examination or inspection.

By direction of the Administrator of Civil Aeronautics:

Refusal to endorse: \_\_\_\_\_  
Civil Aeronautics Inspector

**J. ALLEN McPHEE**  
Civil Aeronautics Inspector

Date: \_\_\_\_\_

**MAY 16, 1945**  
Date of Issuance

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding three years, or both. (over)

*3-44 also*

Inspector's Endorsement	Date

If this certificate is lost or destroyed, a duplicate may be obtained from the Certificate Division Civil Aeronautics Administration, Washington, D. C., for \$1.00 per page. (Money order or check should be made payable to the Treasurer of the United States.)

UNITED STATES OF AMERICA  
Department of Commerce  
Civil Aeronautics Administration  
Washington

## AIRCRAFT INSPECTION REPORT

RUBBER DEVELOPMENT CORPORATION REC. 101 Indiana Avenue N. W. Washington D. C.  
(Owner's name) (Address)  
Grumman aircraft engineering G21-A 37793 46497  
(Make) (Model) (Serial number) (Identification mark)

Designation 7 PC AmFb M Aircraft Specification No. 654

(ISSUANCE) PERIODIC ENDORSEMENT—REPAIR—ALTERATION—SPECIAL. (Circle which

Is Registration Certificate properly displayed in the aircraft? yes

Are Airworthiness Certificate and Operation Record Form in the aircraft? yes

Are aircraft and engine log books available and in proper condition? yes

Periodic aircraft inspection report dated 5/15/45 by P. Nyholm submitted.

1. Aircraft was inspected and determined to have complied with the following, except as noted—\*

(a) Listing in inspector's handbook, chapter XVIII

(b) Aircraft specifications Serial No. A- 654-2

(c) Special inspection request—dated

(d) Notes Nos. 1, 2b.

(e) Special Notes Nos. none pertinent

\* Exceptions, if any, noted

Satisfactory? Yes (XX). No ( ).  
2. Aircraft was thoroughly inspected and determined to be in an airworthy condition for the operations authorized and for the type of certificate issued, except as noted.

\* Exceptions, if any, of which the owner or his representative was advised

Satisfactory? Yes (XX). No ( ).  
3. Alterations and/or major repairs described in the attached Repair and Alteration Form were thoroughly inspected and determined to be satisfactory, and to comply with the provisions of the Civil Air Regulations pertaining to repair and alteration, except as noted.

\* Exceptions, if any, of which the owner or his representative was advised

Satisfactory? Yes (XX). No ( ).  
4. All required placards, instructions, warnings, and markings pertaining to the use and operation of the controls, fuel valves, instruments and gages, safety devices, fuel, oil and baggage capacities, and the like, are correctly executed and properly displayed, except as noted.

\* Exceptions, if any, of which the owner or his representative was advised

Satisfactory? Yes (XX). No ( ).  
5. Based on the results of the above inspection, Aircraft Airworthiness Certificate with Operation Record Form attached, was issued (XX). Endorsed ( ) for a period of on date of

DISAPPROVED \*

APPROVED

Date

Date May 16, 1945

Inspector

Inspector Allen McPeck

[OVER]

REMARKS PERTAINING TO—

AIRCRAFT Total time 120 hrs.

ENGINES 120 hrs. since new and no overhaul.

PROPELLERS 120 Hrs since new- no overhaul

REPAIRS none involved

After accident of none

ALTERATIONS none involved

Forms and documents submitted (indicate by form number or nature of content) which are attached Form ACA 132, 501, cc of 501-A, cc of letter from W. W. Brinkerhoff 3/30/45, Form 305, cc of 303-A, Form 309 cc of pages 1, 2, 3, and 4, and form 307, cc of telegram to (1-270) signed 'Lanter' 5-5-45 with the following to be checked by the Aircraft Airworthiness Section

and returned when checked ( ). Forwarded through channels if satisfactory (XX).

REMARKS This aircraft was given a gross load of 8800 pounds upon the basis described in the ~~attached~~ above listed telegram.

RECEIVED  
CERTIFICATE SECTION

NOTE.—This form to be executed and submitted in single copy following any inspection which results in the issuance or endorsement of an Aircraft Airworthiness Certificate, as: (1) Original issuance, (2) periodic endorsement, or (3) following major repair, alteration, or special inspection, except that in case of disapproval, a duplicate copy of the "Disapproved" Inspection Report will be attached to the Aircraft Operation Record, and will only be removed by an inspector of the ADMINISTRATION, FOLLOWING A SATISFACTORY INSPECTION OF THE AIRCRAFT.

A26759

UNITED STATES OF AMERICA  
CIVIL AERONAUTICS AUTHORITY  
WASHINGTON, D. C.

IDENTIFICATION MARK <u>46497</u>
DATE <u>5-16-45</u>
APPROVED <i>[Signature]</i> (Inspector's signature)
DISAPPROVED  (Inspector's signature)

APPLICATION FOR AIRWORTHINESS CERTIFICATE FOR AN AIRCRAFT

To the CIVIL AERONAUTICS AUTHORITY:

Application is made for an Airworthiness Certificate for:

1. Make Grunman
2. Model G-21A--(JRF5)  
37793
3. Type Certificate No. 454
4. Manufacturer's Serial No. 37793
5. Applicant's name Rubber Development Corporation  
(Print or type FULL NAME)
6. Permanent mailing address 101 Indiana Ave., N.W. Washington D.C.  
(Street) (City) (State)
7. Describe any special purposes for which aircraft will be used  
Crude rubber procurement program in S.A.

The aircraft is now registered with the Authority: }  
Application for registration is attached hereto: } in the name of the applicant.

I CERTIFY that the above statements are true.

Rubber Development Corp.  
(Name of applicant)  
By Chas D Bailey  
(Signature)  
Chief, Aviation Div.  
(Title)

DATE May 15, 1945

1004  
1005



To: 1-315 (Mr. McPeeke)

CIVIL AERONAUTICS ADMINISTRATION

SAFETY REGULATION

ESR-ACA-1945

*Cop 1*

MAY 5, 1945 4:10 P. M.

Washington, D. C.

WESTERN UNION - DAY LETTER

Civil Aeronautics Administration (1-270)  
HFZ New York, New York

REPHONE VOLLMECKE. RUBBER DEVELOPMENT CORPORATION GRUMMAN MODEL G-21.  
SATISFACTORY TO CERTIFICATE ON NR BASIS AT 8,800 POUNDS IN ACCORDANCE  
WITH FOLLOWING RESTRICTIONS. ONE OPERATE FROM SHELTERED WATER OR  
PREPARED LANDING AREAS AS PER NAVY ORDER 96-42 TWO USE ONLY FOR CARGO  
AND OR CARRIAGE OF EMPLOYEES OF RDC AND CREW ESSENTIAL TO PURPOSES OF  
FLIGHT.

Lanter

RDHoekstra:dbb

cc: A-270, A-297, 1-290, 1-297

*Rec'd 1-315 (M) 5-18-45*

10-10-1968



UNITED STATES OF AMERICA  
DEPARTMENT OF COMMERCE  
CIVIL AERONAUTICS ADMINISTRATION  
WASHINGTON

APPLICATION FOR FOREIGN FLIGHT AUTHORIZATION

(To be typewritten and filed in duplicate) NOTE. Aircraft will also be operated between Manaus, Brazil, and Bolivia, Colombia, and Peru. fjb

To the Civil Aeronautics Administration:

Application is made for a Foreign Flight Authorization pursuant to section 60.94 of the Civil Air Regulations for a flight to Manaos, Brasil leaving the United States on or about May 5, 1945, and to return on or about May 1, 1946.

1. Purpose of the flight (tourist, delivery, commercial, sales, etc.): Ferry Aircraft assigned to our activities in South America. Aircraft is to be operated in Brazil. See

Name and address of financial sponsor of the flight Rubber Development Corporation (above)  
101 Indiana Ave., Washington, D.C.

Name of foreign country in which aircraft will remain permanently

2. Aircraft: Registered owner Rubber Development Corporation

Address 101 Indiana Ave., Washington, D.C.

Manufacturer Grunman

Model G-21-A - JRF-5

Serial No. 37793

Identification mark 46497 - NC

Expiration date of Aircraft Airworthiness Certificate

3. Engines:

Manufacturer and model

Serial No.

H. P.

Pratt & Whitney - R-985 Wasp

JP 215011

400

(Made by Jacobs)

JP 215012

4. Radio: State call letters assigned transmitter - TA-12B Describe radio equipment installed:

Receiver - RA-1B with loop

5. Special apparatus, equipment, or cargo (description of the following):

Cameras None

Firearms None

Military equipment None

Cargo (except personal baggage) Aircraft spare parts and/or aviation equipment.

Will cameras or firearms be sealed during flight outside the jurisdiction of the United States? If not, state in what countries request is made to take aerial photographs

6. Flight crew:

Pilot

Copilot & Flight

Mechanic

Name Frank L. Sage John Ganz Eng.

Address Washington, D.C. Washington, D.C.

Citizenship U.S.A. U.S.A.

Class of certificate Commercial Private-Mechanic

Expiration date of certificate

Certificate No. 37611 57223-M-2861

Ratings held S.&M. Engine -A&E

L & S-275/3500 H P Last medical

8-8-44. fjb

12-15-43

fjb

Name. . . . .  
 Address. . . . .  
 Citizenship. . . . .  
 Class of certificate. . . . .  
 Expiration date of certificate. . . . .  
 Certificate No. . . . .  
 Ratings held. . . . .

## 7. Passengers:

Name and address

Citizenship

None on flight to Brazil. The passengers carried in South America will be limited to representatives of the Rubber Development Corp. and/or individuals flying in the interest of the Rubber Development Corporation.

8. Liability insurance: Name of company **Olen Falls, Ind.**Date policy issued **January 1,** 194**5** Date expires **January 1,** 194**6**Amount of coverage: Property **100/300,00** Persons **50/300,000**

## Itinerary and landing points:

City	Country	Airport*	Date of arrival	Date of departure
Miami, Florida	U.S.A.	36th Street		5/5
Camaguey	Cuba	Gen. I. Agramonte	5/5	5/5
Port of Prince	Haiti	Bowen Field	5/5	5/5
San Juan	Puerto Rico	Isla Grande	5/5	5/6
Antigua	B.W.I.	Coolidge	5/6	5/6
Port au Spain	Trinidad	Piarco	5/6	5/7
Georgetown	B. Guiana	Atkinson	5/7	5/7
Cayenne	F. Guiana	Rochambeau	5/7	5/7
Belem	Brasil	Val de Caes	5/7	5/7
Santarem	Brasil	Mapine	5/8	5/8
Manaos			5/8	5/8

(Attach additional sheets as required)

\*Indicate by the symbols E and C, the airports in each country, colony, or possession at which entry (E) and clearance (C) requirements will be complied with

## 10. Does route to be flown involve an extensive, difficult, or long over water flight? If so, submit the following:

- A schedule giving distances between landing points together with map of route.
- Data which will show that the aircraft is capable of completing flight between landing points. This should include, cruising speed to be used; gross weight of the aircraft at take-off; necessary power output; length of runways.
- Statement of navigation methods and equipment that will be used.

## 11. Additional information: State such other information and data as may be necessary and appropriate in order to set forth fully the particular circumstances of the proposed foreign flight

Re Brazil, Colombia, Bolivia and Peru. It is understood that flights of the aircraft in these countries will be made in accordance with the provisions of the air agreements in force between the governments of these countries and the United States Government. fjb

## 12. Person to whom foreign flight authorization should be mailed:

Name and address **Charles D. Bailey - Rubber Development Corporation**  
**101 Indiana Ave., Washington, D.C.**

The foregoing information is, to the best of my knowledge and belief, correct and complete.

**RUBBER DEVELOPMENT CORPORATION**

By *[Signature]*

(Registered owner of the aircraft)

**Chief, Aviation Division**

Date **April 23,** 194**5**

(Pilot in command of the flight)

DEPT. OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 <b>FOR FAA USE ONLY</b> OFFICE IDENTIFICATION 50-FSDO-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>	MODEL <b>G-21A</b>		NATIONALITY AND REGISTRATION MARK <b>N1098V</b>	
	SERIAL NO. <b>37293</b>				
2. OWNER	NAME (As shown on registration certificate) <b>ANTILLES AIR BOATS</b>		ADDRESS (As shown on registration certificate) <b>WEST SEAPLANE RAMP CHRISTIANSTED ST. CROIX U.S.V.I. 00820</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				
POWERPLANT	<b>Pratt &amp; Whitney</b>	<b>R-985 ANI-I4B</b>	<b>JP 207131</b>	<b>X</b>	
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
PART 43 CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		
<b>Antilles Air Boat, Inc.</b> <b>Isla Grande Airport</b> <b>San Juan, Puerto Rico. 00927</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		
			C. CERTIFICATE NO. <b>A&amp;P 52858</b>		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>June 26-1979</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>George S. Martin</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION <input checked="" type="checkbox"/>		OTHER (Specify) <b>IA.</b>
	FAA DESIGNEE	REPAIR STATION			
DATE OF APPROVAL OR REJECTION <b>June 26-1979</b>		CERTIFICATE OR DESIGNATION NO. <b>52858</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <i>George S. Martin</i>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

This engine, received at Antilles Engine Overhaul Shop, for a complete major overhaul. Engine received a Pratt & Whitney Model R-985 ANI-I4B. Engine complete and no external damage found in this engine. Engine given complete dissassembly. All steel parts cleaned and inspected for proper clearances. All steel parts been magnaflux by Arnold Greene Testing Lab., of Puerto Rico. FAA Repair Sta. 761-I. Invoice no. 5858. At 2/16/79. All aluminum parts cleaned and insp., through ZEGLO check. New ignition harness installed. Install overhaul magnetos. After the assembling, engine been run at test cell for 3hrs. All this repair been performed through the Pratt & Whitney Overhaul Manual. The following AD, been C/W and Previous insp. Engine records at the overhaul shop file.

56-6-2 C/W.

57-05-04 PC/W.

66-14-4 PC/W.

68-9-1 C/W.

78-08-07 C/W. Part B.

NATIONALITY		REGISTRATION MARK		DATE	
LIBERIA		N-302 VAI-IND		23 SOAINT	
DESCRIPTION OF WORK ACCOMPLISHED					
<p>56-6-2 C/W.</p> <p>57-05-04 PC/W.</p> <p>66-14-4 PC/W.</p> <p>68-9-1 C/W.</p> <p>78-08-07 C/W. Part B.</p>					
<p>ADDITIONAL SHEETS ARE ATTACHED</p>					

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)		Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>50-FSD0-61</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.			
1. AIRCRAFT	MAKE <b>GERMAN</b>	MODEL <b>G-21A</b>	
	SERIAL NO. <b>37793</b>	NATIONALITY AND REGISTRATION MARK <b>N1048V</b>	
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc.</b>		ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp Christiansted, St. Croix, USVI 00820</b>
	3. FOR FAA USE ONLY		
4. UNIT IDENTIFICATION			
UNIT	MAKE	MODEL	SERIAL NO.
AIRFRAME	***** (As described in item 1 above) *****		<b>X</b>
POWERPLANT			
PROPELLER			
APPLIANCE	TYPE		
	MANUFACTURER		
6. CONFORMITY STATEMENT			
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY	C. CERTIFICATE NO.
<b>ADALBERTO AGOSTO</b> <b>P.O. Box 20865</b> <b>Rio Piedras, P. R. 00928</b>		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC	<b>A &amp; P 1727145</b>
		<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC	
		<input type="checkbox"/> CERTIFICATED REPAIR STATION	
		<input type="checkbox"/> MANUFACTURER	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.			
DATE <b>June 11, 1979</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Adalberto Agosto</i> <b>ADALBERTO AGOSTO</b>	
7. APPROVAL FOR RETURN TO SERVICE			
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED			
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION <input checked="" type="checkbox"/> CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	
DATE OF APPROVAL OR REJECTION <b>June 11, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>IA1617154</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <b>CHARLES FREHLING</b> <i>Charles Frehling</i>

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed main landing gear wheels (two) Hayes Model #G3-56A 30X7 together with brake assemblies (two) Hayes Model H-2-102.
2. Installation of Cleveland wheel and brakes supplemental type Certificate Number SA99GL Dated December 3, 1975.
3. Installed Cleveland wheel and brake conversion kit P/N 199-65 in accordance with Cleveland installation drawing 50-25 Revision A dated December 12, 1978, consisting of wheel assembly P/N 40-127 and Brake assembly P/N 30-97.
4. Aircraft weight and equipment list reflect this alteration.

----- NOTHING FOLLOWS -----

ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-P500-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>		MODEL <b>G-21A</b>		
	SERIAL NO. <b>57793</b>		NATIONALITY AND REGISTRATION MARK <b>N1048V</b>		
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc.</b>		ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp Christiansted, St. Croix, USVI 00820</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
<b>ADALBERTO AGOSTO</b> <b>P.O. Box 20865</b> <b>Rio Piedras, P.R. 00928</b>		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>A &amp; P 1727145</b>	
		<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC			
		<input type="checkbox"/> CERTIFICATED REPAIR STATION			
		<input type="checkbox"/> MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>June 11, 1979</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>ADALBERTO AGOSTO</b> <i>Adalberto Agosto</i>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	OTHER (Specify)		
	FAA DESIGNEE	REPAIR STATION			
		<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION			
		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT			
DATE OF APPROVAL OR REJECTION <b>June 11, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>IA1617154</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>CHARLES FRIEHLING</b> <i>Charles Friebling</i>	

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Supersedes previous radio installation of July 11, 1973.
2. Removed two (2) each Genave Nav./Com. transceivers Model Alpha 200 at Station #9, right side on rack fastened to cockpit overhead to existing structure.
3. Installed one (1) Escort Nav/Com. transceiver Model H110 per Narco installation Manual and installed one (1) Bendix ADF T-12C Nav Receiver, installed in accordance with Bendix Installation Manual.

Both installed per AC43-13-2, Chapter 2, Section 27.

See Weight and Balance Records and Equipment List and Electrical Load Analysis.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED



DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION <b>50-FSDO-61</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>			MODEL <b>G-21A</b>	
	SERIAL NO. <b>37793</b>			NATIONALITY AND REGISTRATION MARK <b>N1048V</b>	
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc.</b>			ADDRESS (As shown on registration certificate) <b>West Soaplane Ramp Christiansted, St. Croix, USVI 00620</b>	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	(As described in item 1 above)			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>ADALBERTO AGOSTO</b> <b>P.O. Box 20865</b> <b>Rio Piedras, P.R. 00928</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>A &amp; P 1727145</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>June 11, 1979</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Adalberto Agosto</i> <b>ADALBERTO AGOSTO</b>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	<input type="checkbox"/>		
		INSPECTION AUTHORIZATION			
		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT			
DATE OF APPROVAL OR REJECTION <b>June 11, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>IA1617154</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles Fuesling</i> <b>CHARLES FUESLING</b>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Installed Alcor Inc. Exhaust gas temperature indicator kit #326 I/A/W supplemental type certificate Number SA96SW per report 68-145 F.A.A. approved list dated 4/26/77.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-F800-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>	MODEL <b>G-21A</b>		NATIONALITY AND REGISTRATION MARK <b>N1048V</b>	
	SERIAL NO. <b>37793</b>				
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc.</b>		ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp Christiansted, St. Croix, USVI 00820</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
				REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				<b>X</b>
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
<b>ADALBERTO AGOSTO P.O. Box 20865 Rio Piedras, P.R. 00928</b>		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		<b>A &amp; P 1727145</b>	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>June 11, 1979</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>ADALBERTO AGOSTO</b> <i>Adalberto Agosto</i>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>June 11, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>1A1617154</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <b>CHARLES FRIEDLING</b> <i>Charles Friedling</i>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Supersedes previous Hartzell Propeller installation of July 11, 1973.
2. Installation of Hartzell Propeller supplement type Certificate Number SA-52 Dated January 19, 1967.
3. Installed (two) each Hartzell Propeller Model HC-B3R30-2B, with blade model R10152-S<sub>2</sub> in accordance with Hartzell drawing #D1860.
4. Installed emergency governor cross feed system for unfeathering per Alaska Coastal Airline Drawing List Dated March 8, 1957.
5. Aircraft weight and equipment list reflect this alteration.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>50-FS00-61</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	GUDMAN		MODEL	G-21A
	SERIAL NO.	37793		NATIONALITY AND REGISTRATION MARK	N1048V
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	Antilles Air Boats, Inc.			West Shaplane Ramp Christiansted, St. Croix, USVI 00820	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
				X	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
ADALBERTO AGOSTO P.O. Box 20865 Rio Piedras, P.R. 00928			X		A & P 1727145
			U.S. CERTIFICATED MECHANIC		
			FOREIGN CERTIFICATED MECHANIC		
			CERTIFICATED REPAIR STATION		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE			SIGNATURE OF AUTHORIZED INDIVIDUAL		
June 11, 1979			ADALBERTO AGOSTO <i>Adalberto Agosto</i>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	X	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
June 11, 1979		1A1617154	CHARLES FRIEDBERG <i>Charles Friedberg</i>		

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed corroded front top box beam capstrip from station 40 left to station 37 right, installed new box beam. Center section box beam capstrip insertion repair left and right, work done in accordance with Section 2, Figure 14 NAV.AER. 01-85V-3.
  2. Removed left wing fuel tank bulkhead station #30, see figure #10 wing center section box beam structure. Original bulkhead 24St .051 replaced with 2024-T3.051. All gas tightjoints per Par 91 and 92. NAV.AER. 01-85VA-3.
  3. Removed corroded rear top box beam capstrip from station 15 right to station 42 right installed new box beam. Center section Box beam capstrip insertion repair, work done in accordance with Section 2, Figure 14.
  4. Removed corroded rear bottom box beam capstrip station 15 left to station 54 left installed new box beam. Center section box beam capstrip insertion repair, work done in accordance with Section 2, Figure 14.
  5. Removed corroded sections of 2 members Ref. Figure #10, Item #8 station 30 left to station 30 right top and bottom, repair carried out in accordance with center section Figure 17 - Wing Box Beam Stringer Repairs Section 2 NAV.AER. 01-85V-3.
  6. Removed corroded top center section skin Ref. Figure 4 Skin # 32 wing skin plating diagram station 44 left to station 40 right, installed new skin 2024-T3.064, skin splice plate at station 44 left makes up repair plate in item 1 box beam repair. Installed skin splice plate Rt. at station 44.
  7. Replaced leading edge skin Lt wing skin #18, station 226 to station 241. Original skin 24ST.028 replaced with 2024-T3.032.
- A. All work done in accordance with Grumman Structure and Repair Manual NAV.AER. 01-85V-3, and AC43-13-1 Chapter 2.

NOTHING FOLLOWS

ADDITIONAL SHEETS ARE ATTACHED

MAJOR REPAIR AND ALTERATION  
(Airframe, Powerplant, Propeller, or Appliance)FOR FAA USE ONLY  
OFFICE IDENTIFICATION  
50-P500-61

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 for subsequent revision thereof for instructions and disposition of this form.

1. AIRCRAFT	MAKE <b>GRUMMAN</b>	MODEL <b>G-21A</b>
	SERIAL NO. <b>37793</b>	NATIONALITY AND REGISTRATION MARK <b>N1048V</b>
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc.</b>	ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp Christiansted, St. Croix, USVI 00820</b>

## 3. FOR FAA USE ONLY

## 4. UNIT IDENTIFICATION

## 5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	As described in item 1 above				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE	MANUFACTURER			

## 6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Abelaro Gomez C/O Antilles Air Boats, Inc. West Seaplane Ramp Christiansted, St. Croix, USVI 00820	<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC	A & P 1736931
	<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC	
	<input type="checkbox"/> CERTIFICATED REPAIR STATION	
	<input type="checkbox"/> MANUFACTURER	

D. I certify that the repair and or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE <b>June 8, 1979</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <b>Abelaro Gomez</b> <i>Abelardo Gomez Ramos</i>
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## 7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED.

BY	FAA ST. STANDARDS INSPECTOR	FAA DESIGNEE	INSPECTION AUTHORIZATION	OTHER (Specify)
			<input checked="" type="checkbox"/> CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION	CERTIFICATE OF DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
<b>June 8, 1979</b>	<b>TA 1617153</b>	<b>CHARLES FREEHLING</b> <i>Charles Freehling</i>		

## NOTICE

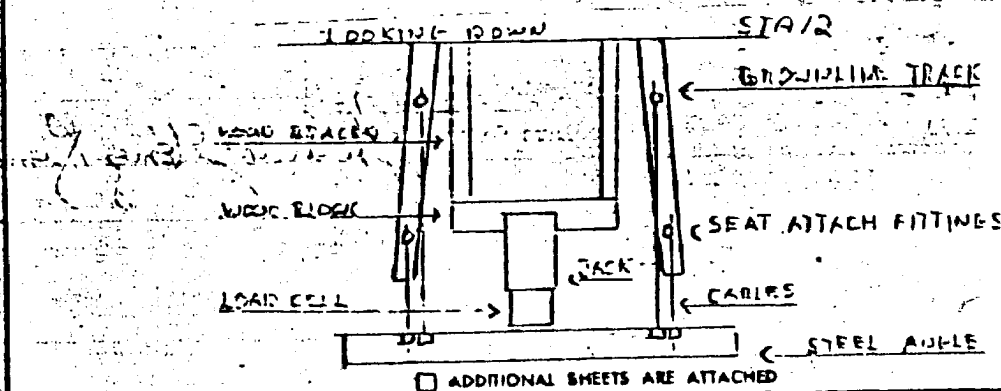
Weight and balance or other limitation changes shall be entered in the appropriate aircraft records. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

B. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- A. Installation of new associated Aircraft Equipment, Inc. Pilot seat model G21-P TSOC39A Type 11. Brownline track attached to floor as follows.
1. Each track installed on floor using 6, 3/16 AN 509 Screw; 5 each #10 and one each 1/4.
  2. The first two screws on each side installed through the track, floor and into a 1" x 1" x 1/8 2024T3 angle.
  3. The 3rd screw on each side installed through the track, floor and into the angle which is attached to station 11.
  4. The 4th and 5th AN 509 screw installed into same support beam as original rear chair legs.
  5. The 6th screw installed through track and into angle which is part of station #12.
- B. The test carried out for the installation of the Brownline track attachment to the aircraft structure as per AC 43-13-2 Chapter 1, Structural Data, Static Test, Figure 1.1 and part 23.561 General. All work done in accordance with AC 43-13-1A Chapter 2.

The Static Load Test performed as follows on each Brownline track which was installed in the same location as original pilot seat. Seat attach fittings were installed same location as seat legs to simulate installation. Attached to each fitting was a 3/16" cable pulled forward and up. The average angle of all four cables was 45°, the cables were attached to a steel angle. A block of wood cut on an angle and positioned on the floor and braced to station #12. On this block, a 1 1/2 ton hydraulic jack with scale load cell was placed against the steel angle, apply load of 2,200 lbs. was demonstrated. No damage or permanent deformation occurred after about 10 seconds on right side. On left side, front forward, the floor showed little deformation which, after the load was released, went back to normal configuration. After the test all structures were inspected and found airworthy.

- C. Installed Bon Corporation P/N E2884-3 shoulder harness. Attach point at Station #13 using 1" x 1" angle .125" thick; two pieces backing each other 14" long; riveted using AN470-4 rivets single pattern 1" spacing attached to cockpit entrance door frame and bulkhead stiffener; 6" from top of left main gear well. Shoulder Harness attached to angles with AN5-11 bolts with AN365-24 lock nut. This is the same installation as in N74588, December 7, 1978.





DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>50-FSDO-61</b>	
<b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)					
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>CELESTAN</b>			MODEL <b>G-21A</b>	
	SERIAL NO. <b>37793</b>			NATIONALITY AND REGISTRATION MARK <b>N1048V</b>	
2. OWNER	NAME (As shown on registration certificate) <b>ANTILLAS AIR BOATS, INC.</b>			ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp, Christiansted St. Croix, U.S.V.I. 00820</b>	
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT				X	X
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>ADALBERTO AGOSTO</b> P.O. BOX 20865 RIO PIEDRAS, P.R. 00920			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>A &amp; P 1727145</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>June 4, 1979</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Adalberto Agosto</i> <b>ADALBERTO AGOSTO</b>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>June 4, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>IA 1617154</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>CHARLES R. FREELING</b> <i>Charles R. Freeling</i>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Replaced hull skin #15, #16, and #17 left side from chine to watertight line as seen on Figure #62 new skin 202473 .032, splice skins as per Figure #63 - hull skin insertion repair.
- (2) Installed new left, center and right windshield post center at station #8 and left and right station #9. Installed new windshield left and right.
- (3) Repaired aft hull left side station 17 $\frac{1}{2}$  to station 27 $\frac{1}{2}$ , 6" up from center of keel and right side station 23 $\frac{1}{2}$  to station 25 $\frac{1}{2}$ , 6" up from center of keel, repair as per Figure #65 - Hull bottom repair - Flush type.
- (4) Removed all existing seats and flooring, Carrier Aircraft Installation. FAA Form 337, July 11, 1973.
  - (a) Reinstalled two each lounges. See Drawings attached to 337, 7/11/73.
  - (b) Reinstalled four each seats. See Drawings attached to 337, 7/11/73.
  - (c) The above mentioned seats and lounges were attached to Structure Substantiation Report dated 15, June, 1969, #621-1 approved by DER LA-321 affixed to 337, 7/11/73. Minor changes to this installation which do not affect the structural integrity of the installation, are as follows:  
On each transverse floor beam, an angle was installed by riveting, which makes it possible to use four #5 AN470 rivets for attachments of seat channels to the floor beam on each transverse. Also in place of the Douglas Fir Plywood floor, G11floor 5007A was substituted, see attached report.
  - (d) This is the same installation as G-21A, N74588 FAA Form 337, 11/17/78.
- (5) All work above done in accordance with Grumman structure and repair manual NAV. AER. 01-85V-3, and A643-15-1A, Chapter 2, Section 3.

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.				OFFICE IDENTIFICATION 50-P500-61	
1. AIRCRAFT	MAKE GILSON	SERIAL NO. 37793	MODEL G-21A	NATIONALITY AND REGISTRATION MARK US N1048V	
2. OWNER	NAME (As shown on registration certificate) ANTILLES AIR BOATS, INC.		ADDRESS (As shown on registration certificate) West Seaplane Ramp, Christmsted St Croix, U.S.V.I. 00820		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)			X	
POWERPLANT					
PROPELLER					
APPLIANCE					
5. TYPE					
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS ADELBERTO AGOSTO Antilles Air Boats, Inc. San Juan, Puerto Rico			B. KIND OF AGENCY <input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		C. CERTIFICATE NO. A & P 1727145
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE May 24, 1979		SIGNATURE OF AUTHORIZED INDIVIDUAL ADELBERTO AGOSTO			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY:	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION		
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION May 24, 1979		CERTIFICATE OR DESIGNATION NO. IA 161714		SIGNATURE OF AUTHORIZED INDIVIDUAL Charles A. Kneeling	

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Removed damaged nose Station -87/8 to Station 9 end of skin 5, 4 and 32 and Station 11 1/2 end of skin 6. Replaced the following skins #1 nose bumping, #2 left and right, original skin, .028 334H, new skin 2024T3 .032. #3 deck station 3 to 6 original skin .028 334H new skin 2024T3 .032. #4 left and right upper side Station 3 to Station 9 original skin 24 st .028 replaced by 2024T3 .032. #5 deck station 6 to station 9 original skin 24 st replaced by .032 replaced by 2024T3 .032, #6 left and right lower side Station 1 to Station 11 1/2 original skin 24 st .028 replaced by 2024T3 .032, #32 left and right middle side Station 1 to Station 9 original skin 24 st .028 replaced by 2024T3 .032. #29 left and right used original Grumman skin. #30 left and right, original skin 24 st .057 replaced by 2024T3 .063 splice at Station 6 1/2. Splice Plate .063 edge up and two runs rivet each side, splice right side Station 6 1/2 and 7 plate .063 Edges up and eight rows of rivets. Keel spliced I/A/W Figure 67, Nav. Aer. 01-85V-3. Chine replaced left and right Station 1 to Station 11 1/2. Main bulkhead Station 1 and bulkhead Station 6, vertical frames, cross members, flooring, angle frames, stringers and horizontal stiffeners either Grumman parts or repaired I/A/W Nav. Aer. 01-85V-3 Section 4, Hull Par. 1 to 72. All joints forward of Station 9 are watertight and skin joints are watertight as shown in Figure 62, which also notes watertight bulkheads.
- (2) Replaced panel at Station 9, see Figure #71 Hull Structure, Pilot's Cabin Ref. No. 5, Panel 24st. .064 replaced by 2024T3 .071 anodized.
- (3) Flight Instrument Panel modified to provide standardization throughout the fleet. Standardization of layout allows more efficient scanning of instruments by pilot and also permits improved arrangement of pitot, static and vacuum lines.  
A. Panel material: 24st. 070 replaced by 2024T3. 125 anodized. Mounting: Original G-21A location and shock mounts.
- (4) Replaced pilot's cabin skin, top left and right station #9.5 to Station 12, Ref. Figure 62, Hull Skin Plating Diagram Skin #39.
- (5) Repaired upper side skin #8 from Station 9 1/2 to 12 original skin .028 replaced by 2024T3 .032 splice at Station I/A/W Fig. 63 Hull Skin Insertion.
- (6) Replaced Hull Skin #7 left side Station 9 to 14, ref. Figure #62, Hull Skin Plating Diagram, Skin 2024T3 .040.
- (7) Replaced Hull Skin #7 right side Station 9 to 14, ref. Figure #62, Hull Skin Plating Diagram, Skin 2024T3 .040.
- (8) Replaced right dome side - outer skin #39, ref. Figure #62, Station 9.5 to Station 12, new skin 2024T3 .032.
- (9) Replaced Hull Skin #12 left side Original Skin 24 st .028, new skin 2024T3 .032 Ref. Figure #62.
- (10) Replaced Hull Skin #10 left side original skin 24 st .028, new skin 2024T3 .032, Ref. Figure #62.
- (11) Replaced Hull Skin #14 under cabin door Station 24 to Station 26, New Skin 2024T3 .040. Ref. Figure #62.
- (12) All work done in accordance with Grumman Structure and Repair Manual, Nav. Aer. 01-85V-3, and AC 43-13-1A, Chapter 2, Section 3, and Chapter 3, Section 3.

NOTHING FOLLOWS

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-RSD0-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	GERMAN		MODEL	G-21A
	SERIAL NO.	37793		NATIONALITY AND REGISTRATION MARK	US N1048V
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	ANTILLES AIR BOATS, INC			West Seaplane Ramp, Christiansted, St Croix, U.S.V.I. 00830	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	***** (As described in item 1 above) *****				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY.		C. CERTIFICATE NO.	
ALBERTO AGOSTO ANTILLES AIR BOATS, INC SAN JUAN, PUERTO RICO		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		A & P 1727148	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE: May 24, 1979		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Alberto Agosto</i> ALBERTO AGOSTO			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION May 24, 1979		CERTIFICATE OR DESIGNATION NO. IA 1617144		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles A. ...</i> CHARLES A. ...	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) In order to facilitate inspection of rudder and elevator push-pull rods, Bell cranks and Bellcrank attach fittings an inspection access door has been provided as follows:
- A. One hole made on each side of the hull between hull stations nine and ten, 16.5 inches above the chine, 14.5 inches forward of each main gear wheel well. All corners 5/16" radius.
  - B. One doubler of 2024T3 X.050 X 10 3/4" X 9 3/4" riveted on the inside of the hole to restore skin strength at the location of the access hole. Double row of rivets used with 1" spacing as in skin joints of airframe.
  - C. A hole of 54" vertical dimension and 6.5" longitudinal dimension has been cut in the doubler and provides actual inspection facilities as mentioned above.
  - D. 88 x 32 gang-nut channel with 3/4" spacing provides the attachment of watertight cover constructed in the same manner as watertight doors on floats and large watertight door on wheelwell on R/H side of hull. Cover installed on sealant surface and torqued in same manner as wheel well inspection door.
- (2) Installed Grimes single strobe light, Part Number 30-0457-1, under the provisions of FAA Advisory Circular 20-30A and AC 43-13-1A, Chapter 11, Section 2, 3 and 7. Installed atop the vertical stabilizer in accordance with AC 43-13-2, Chapter 4, Paragraph 45-D, utilizing 20 gauge wire through 5 amp circuit breaker. Light demands 1 amp at 28 volts 10C - unit weight, 2 lbs.

Paragraphs 1 and 2:

All work done in accordance with Grumman Structure and Repair Manual Nav. Aer. 01-85V-3, page 196, Figure 110 and AC 43-13-1, Chapter 2, Section 3, paragraph 60-68. AC-43-13-2, Chapter 1, Paragraphs 4 thru 8.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-FSD0-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE GRUMMAN		MODEL G-21A		
	SERIAL NO. 37793		NATIONALITY AND REGISTRATION MARK US N1048V		
2. OWNER	NAME (As shown on registration certificate) ANTILLES AIR BOATS, INC		ADDRESS (As shown on registration certificate) West Seaplane Ramp, Christiansted, St. Croix, U.S.V.I. 00820		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
ADELBERTO AGOSTO ANTILLES AIR BOATS, INC SAN JUAN, PUERTO RICO		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		A & P 1727145	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE May 24, 1979		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Adelberto Agosto</i> ADELBERTO AGOSTO			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION May 24, 1979		CERTIFICATE OR DESIGNATION NO. IA 1617154		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles R. Fitchling</i> CHARLES R. FITCHLING	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)**

- (1) In order to facilitate inspection of rudder and elevator push-pull rods, Bell cranks and Bellcrank attach fittings an inspection access door has been provided as follows:
- A. One hole made on each side of the hull between hull stations nine and ten, 16.5 inches above the chine, 14.5 inches forward of each main gear wheel well. All corners 5/16" radius.
  - B. One doubler of 2024T3 X.050 X 10 3/4" X 9 3/4" riveted on the inside of the hole to restore skin strength at the location of the access hole. Double row of rivets used with 1" spacing as in skin joints of airframe.
  - C. A hole of 5 1/4" vertical dimension and 6.5" longitudinal dimension has been cut in the doubler and provides actual inspection facilities as mentioned above.
  - D. #8 x 32 gang-nut channel with 3/4" spacing provides the attachment of watertight cover constructed in the same manner as watertight doors on floats and large watertight door on wheelwell on R/H side of hull. Cover installed on sealant surface and torqued in same manner as wheel well inspection door.
- (2) Installed Grimes single strobe light, Part Number 30-0437-1, under the provisions of FAA Advisory Circular 20-30A and AC 43-13-1A, Chapter 11, Section 2, 3 and 7. Installed atop the vertical stabilizer in accordance with AC 43-13-2, Chapter 4, Paragraph 45-D, utilizing 20 gauge wire through 5 amp circuit breaker. Light demands 1 amp at 28 volts 10C - unit weight, 2 lbs.

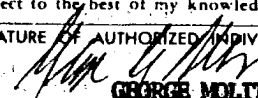
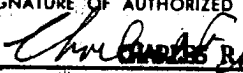
Paragraphs 1 and 2:

All work done in accordance with Grumman Structure and Repair Manual Nav. Aer. 01-85V-3, page 196, Figure 110 and AC 43-13-1, Chapter 2, Section 3, paragraph 60-68. AC-43-13-2, Chapter 1, Paragraphs 4 thru 8.

NOTHING FOLLOWS

☐ ADDITIONAL SHEETS ARE ATTACHED



DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION  <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>50-FS-00-61</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	GUDMAN		MODEL	G-21A
	SERIAL NO.	57793		NATIONALITY AND REGISTRATION MARK	US N1048V
2. OWNER	NAME (As shown on registration certificate)			ADDRESS (As shown on registration certificate)	
	Antilles Air Boats, Inc.			West Seaplane Ramp, Christiansted, St Croix, U.S.V.I. 00820	
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
				X	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
George Molitor c/o Antilles Air Boats, Inc West Seaplane Ramp, Christiansted, St Croix, U.S.V.I. 00820		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>A &amp; P 555244314</b>	
		<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC			
		<input type="checkbox"/> CERTIFICATED REPAIR STATION			
		<input type="checkbox"/> MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE		SIGNATURE OF AUTHORIZED INDIVIDUAL			
May 23, 1979		 <b>GEORGE MOLITOR</b>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL		
May 23, 1979		IA 1617154	 <b>CHARLES R. FIEDLING</b>		

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Grumman G-21A left and right outer wing panels, old fabric removed from structure which was thoroughly cleaned.
- (2) Structure inspected for mechanical damage, cracks, distortion or corrosion.
- (3) Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 2, Wing Group, Wing Outer Panel, and sprayed with zinc chromate primer.
- (4) Recovered right and left wing from Station 97 to 294 using Grade A fabric and doped as follows:
  - A. Three coats clear butyrate dope.
  - B. Three coats aluminum butyrate dope.
  - C. Three coats white butyrate dope.
- (5) All work performed in accordance with AC 43-13-1A Chapter 3, Section 1, 2 and 3. Also Grumman Structural Repair Manual Section 8, Fabric Repair and attachment and Figure 129 Fabric Covering.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

A/c N1048V

Date: May 23, 1979

Page 2

- (3) E. Control surface balancing procedures Grumman G-21A:  
(1) After re-covering or repair: complied with 12, 13  
and 14, Aileron in tolerance.  
F. Installed left side.
- (4) Item (1) all work performed in accordance with AC 43-13-2  
Chapter 3, Section 1, and 2, and Nav. Aer. 01-85V-3 Section 8,  
Fabric Repair and Attachment, par. 2(4).

----- NOTHING FOLLOWS -----

FAA AIRCRAFT REGISTRY

CAMERA NO. 4 DATE: 3-23-81

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-FSDO-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	GRUMMAN		MODEL	G-21A
	SERIAL NO.	37793		NATIONALITY AND REGISTRATION MARK	N-1048V
2. OWNER	NAME (As shown on registration certificate)		ADDRESS (As shown on registration certificate)		
	ANTILLES AIR BOATS, INC.		West Scapiano Ramp, Christiansted, St. Croix, U.S.V.I. 00820		
3. FOR FAA USE ONLY					
The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR 43, section 43.7. 05/31/79 <i>Charles R. Freeling</i> 50-FSDO-61 DATE Signature of FAA Inspector					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
ADELBERTO AGOSTO		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		AAP 1727145	
ANTILLES AIR BOATS, INC.		<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC			
SAN JUAN, PUERTO RICO		<input type="checkbox"/> CERTIFICATED REPAIR STATION			
		<input type="checkbox"/> MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE		SIGNATURE OF AUTHORIZED INDIVIDUAL			
May 22, 1979		ADELBERTO AGOSTO <i>Adelberto Agosto</i>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	
DATE OF APPROVAL OR REJECTION		CERTIFICATE OR DESIGNATION NO.		SIGNATURE OF AUTHORIZED INDIVIDUAL	
May 22, 1979		LA 1617184		CHARLES R. FREELING <i>Charles R. Freeling</i>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- A. Removal of fuel tank and cap top of wing Station 62 left and right.
- B. Installation of repair of hole left from removal of original fuel tank neck, made small circular patch bolted over hole using same bolts as original on neck.
- C. Installed Shaw Aero Devices, Inc., N. Y. flush mounted fuel cap, P/N 428-2, Spec. MIL-C-7244B.
- D. Installed one of the above in top of left and right fuel-tank (Fig. 3, Station Diagram Wing Station 35) 4 1/2 inches aft of front top spar box beam. The flange for fuel cap fits between first and second "Z" stringers. The top skin of the fuel tank (Fig. 4, Wing Skin Plating Diagram, Ref. No. 32, Top Cover Skin) is 24st., .064 inches. Flange of fuel cap is steel, .030 inches thick, installed as per Fig. 108, Skin Patch Repair of the Fuel Tank Skin in accordance with para. 88b, round flush patch, pitch "B" now is 1 inch on new flange.
- E. Installed safety latch which is held in place by air pressure and can be swung away to facilitate fueling.
- F. All work done in accordance with AC 43-13-1A Chapter 2, Section 3, paragraph 96 to 101, Chapter 6, paragraph 247 to 252 and Grumman Structure and Repair Manual Nav. Aer. 01-851A-3.
- G. This installation was previously approved by FAA Form 337 dated November 17, 1978, Grumman G-21A N-74588.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-PSSD-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>	MODEL <b>G-21A</b>		NATIONALITY AND REGISTRATION MARK <b>N-1048V</b>	
	SERIAL NO. <b>37793</b>				
2. OWNER	NAME (As shown on registration certificate) <b>ANTILLES AIR BOATS, INC.</b>		ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp, Christiansted St. Croix, U.S.V.I. 00820</b>		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
<b>ADALBERTO AGOSTO ANTILLES AIR BOATS, INC. SAN JUAN, PUERTO RICO</b>		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>AAP 1727145</b>	
		<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC			
		<input type="checkbox"/> CERTIFICATED REPAIR STATION			
		<input type="checkbox"/> MANUFACTURER			
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>May 22, 1979</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <b>ADALBERTO AGOSTO</b>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION <b>May 22, 1979</b>	CERTIFICATE OR DESIGNATION NO. <b>IA 1617154</b>	SIGNATURE OF AUTHORIZED INDIVIDUAL <b>CHARLES R. FREHLING</b>			

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Sorm Industries Horizontal Stabilizer forward support fitting bulkhead Station 33, Part Number SI-12093-1, in accordance with Sorm Industries FAA sealed Drawing SI-4000, and SI-12093 in accordance with Supplemental Type Certificate Number SA96NW.

..... NOTHING FOLLOWS .....

☐ ADDITIONAL SHEETS ARE ATTACHED



PARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-FSDO-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE	G-21A		MODEL	
	SERIAL NO.	37743		NATIONALITY AND REGISTRATION MARK	
2. OWNER	NAME (As shown on registration certificate)		ADDRESS (As shown on registration certificate)		
	ANTILLES AIR BOATS INC.		WEST SEAPLANE RAMP C' STED ST. CROIX U.S.V.I. 00820		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE	
AIRFRAME	***** (As described in item 1 above) *****			REPAIR	ALTERATION
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
GEORGE NOLITOR c/o Antilles Air Boats Inc West Seaplane Ramp, Christiansted, St Croix, USVI 00820		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		A & P 553244314	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE March 20, 1979		SIGNATURE OF AUTHORIZED INDIVIDUAL GEORGE NOLITOR			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is: <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION March 20, 1979		CERTIFICATE OR DESIGNATION NO. IA 1617154	SIGNATURE OF AUTHORIZED INDIVIDUAL CHARLES R. FROEHLING		

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Elevator Serial #082 Grumman G-21A old fabric removed from structure which was thoroughly cleaned.
- (2) Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 3, Tail Surfaces, and sprayed with two coats of zinc chromate primer.
- (3) Control surface balancing procedures Grumman G-21A.
  - A. Before re-covering surfaces: Complied with 1, 2, 3, 4, 5, 6, 7, 8, 9 Elevator in Tolerance.
- (4) Re-covered with Grade 'A' fabric and doped as follows:
  - A. Three coats clear butyrate dope
  - B. Three coats aluminum butyrate dope
  - C. Three coats white butyrate dope
- (5) All work performed in accordance with AC 43-13-1A, Chapter 3, Sections 1 and 2, and Nav. Aer. 01-85V-3, Section 8, Fabric Repair and Attachment.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				Form Approved Budget Bureau No. 04-R060.1	
MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				FOR FAA USE ONLY	
				OFFICE IDENTIFICATION 50-FSD-61	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE GRUMMAN	MODEL G-21A		NATIONALITY AND REGISTRATION MARK US N1048V	
	SERIAL NO. 37793				
2. OWNER	NAME (As shown on registration certificate) ANTILLES AIR BOATS, INC.		ADDRESS (As shown on registration certificate) WEST SEAPLANE RAMP C'STED ST. CROIX U.S.V.I. 00820		
3. FOR FAA USE ONLY					
4. UNIT IDENTIFICATION					
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)			XX	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS		B. KIND OF AGENCY		C. CERTIFICATE NO.	
CLAUDE AUSTIN c/o ANTILLES AIR BOATS, INC. CHRISTIANSTED, ST CROIX, USVI		<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC <input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC <input type="checkbox"/> CERTIFICATED REPAIR STATION <input type="checkbox"/> MANUFACTURER		A & P 2227109	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE JANUARY 19, 1979		SIGNATURE OF AUTHORIZED INDIVIDUAL CLAUDE AUSTIN <i>Claude Austin</i>			
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/> INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		
DATE OF APPROVAL OR REJECTION JANUARY 19, 1979		CERTIFICATE OR DESIGNATION NO. IA 1617154		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles R. Frehling</i>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Rudder Serial # 077 Grumman G-21A old fabric removed from structure which was thoroughly cleaned.
- (2) Structure inspected for mechanical damage, cracks, distortion or corrosion.
- (3) Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 3, tail surfaces, and sprayed with two coats of zinc chromate primer.
- (4) Re-covered with grade A fabric and doped as follows:
  - A. Three coats clear Butyrate dope.
  - B. Three coats aluminum Butyrate dope.
  - C. Three coats white Butyrate dope.
- (5) All work performed in accordance with AC 43-13-2 Chapter 3, Section 1, and 2, and Nav. Aer. 01-85V-3 Section 8, Fabric Repair and Attachment, Par. 2(6).

Control Surface Balancing Procedures Grumman G-21A

1. After re-covering or repair:

Complied with 12, 13 and 14, Rudder in Tolerance.

----- NOTHING FOLLOWS -----

☐ ADDITIONAL SHEETS ARE ATTACHED

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION <b>MAJOR REPAIR AND ALTERATION</b> (Airframe, Powerplant, Propeller, or Appliance)				Form Approved Budget Bureau No. 04-R060.1 FOR FAA USE ONLY OFFICE IDENTIFICATION <b>SO-ESD-61</b>	
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.					
1. AIRCRAFT	MAKE <b>GRUMMAN</b>		MODEL <b>G-21A</b>		
	SERIAL NO. <b>37793</b>		NATIONALITY AND REGISTRATION MARK <b>US N1048V</b>		
2. OWNER	NAME (As shown on registration certificate) <b>Antilles Air Boats, Inc</b>		ADDRESS (As shown on registration certificate) <b>West Seaplane Ramp, Christiansted, St Croix, U.S.V.I. 00820</b>		
	3. FOR FAA USE ONLY				
4. UNIT IDENTIFICATION					5. TYPE
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(1) (As described in item 1, above)			X	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE				
	MANUFACTURER				
6. CONFORMITY STATEMENT					
A. AGENCY'S NAME AND ADDRESS			B. KIND OF AGENCY		C. CERTIFICATE NO.
<b>George Molitor c/o Antilles Air Boats, Inc Christiansted, St Croix, USVI</b>			<input checked="" type="checkbox"/> U.S. CERTIFICATED MECHANIC		<b>A &amp; P 555244314</b>
			<input type="checkbox"/> FOREIGN CERTIFICATED MECHANIC		
			<input type="checkbox"/> CERTIFICATED REPAIR STATION		
			<input type="checkbox"/> MANUFACTURER		
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
DATE <b>May 23, 1979</b>			SIGNATURE OF AUTHORIZED INDIVIDUAL <i>George Molitor</i> <b>GEORGE MOLITOR</b>		
7. APPROVAL FOR RETURN TO SERVICE					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	<input checked="" type="checkbox"/>	OTHER (Specify)	
	FAA DESIGNEE	REPAIR STATION	<input type="checkbox"/>		
DATE OF APPROVAL OR REJECTION <b>May 23, 1979</b>		CERTIFICATE OR DESIGNATION NO. <b>IA 1617154</b>		SIGNATURE OF AUTHORIZED INDIVIDUAL <i>Charles E. Freese</i> <b>CHARLES E. FREESE</b>	

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- (1) Aileron Serial #AAB 055 old fabric removed from structure which was thoroughly cleaned.
- A. Structure inspected for mechanical damage, cracks, distortion or corrosion.
  - B. Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 3, Tail Surfaces, and sprayed with two coats of zinc chromate primer.
  - C. Control surface balancing procedures Grumman G-21A:
    - (1) Before recovering surfaces: complied with #1 through #9, Aileron in tolerance.
  - D. Re-covered with Grade A fabric and doped as follows:
    - (1) Three coats clear butyrate dope
    - (2) Three coats aluminum butyrate dope
    - (3) Three coats white butyrate dope
  - E. Control surface balancing procedures Grumman G-21A:
    - (1) After re-covering or repair: complied with 12, 13 and 14, Aileron in tolerance.
  - F. Installed left side.
- (2) Aileron Serial #AAB 054 old fabric removed from structure which was thoroughly cleaned.
- A. Structure inspected for mechanical damage, cracks, distortion or corrosion.
  - B. Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 3, Tail Surfaces, and sprayed with two coats of zinc chromate primer.
  - C. Control surface balancing procedures Grumman G-21A:
    - (1) Before re-covering surfaces: complied with #1 through #9, Aileron in Tolerance.
  - D. Re-covered with Grade A fabric and doped as follows:
    - (1) Three coats clear butyrate dope
    - (2) Three coats aluminum butyrate dope
    - (3) Three coats white butyrate dope
  - E. Control surface balancing procedures Grumman G-21A:
    - (1) After re-covering or repair: complied with 12, 13 and 14, Aileron in Tolerance.
  - F. Installed right side.
- (3) Elevator Serial #AAB 083 old fabric removed from structure which was thoroughly cleaned.
- A. Structure inspected for mechanical damage, cracks, distortion or corrosion.
  - B. Structure repaired as necessary in accordance with Nav. Aer. 01-85V-3, Section 3, Tail Surfaces, and sprayed with two coats of zinc chromate primer.
  - C. Control surface balancing procedures Grumman G-21A:
    - (1) Before re-covering surfaces: complied with #1 through #9, Elevator in Tolerance.
  - D. Re-covered with Grade A fabric and doped as follows:
    - (1) Three coats clear butyrate dope
    - (2) Three coats aluminum butyrate dope
    - (3) Three coats white butyrate dope.

☒ ADDITIONAL SHEETS ARE ATTACHED